

# *The Fabrication of Functionally Graded Energetic Materials Using Twin\_Screw Extrusion*

**Frederick Gallant**

*NSWCIH DIV-Indian Head, MD  
& Dept. of Mechanical Engineering  
University of Maryland*

**Hugh A. Bruck**

*Dept. of Mechanical Engineering  
University of Maryland*

**38<sup>th</sup> Annual Gun,  
Ammunition, and Missiles  
Symposium & Exhibition**

**Monterey, CA**

**24-27 March 2003**





# *Sponsors & Participating Organizations*





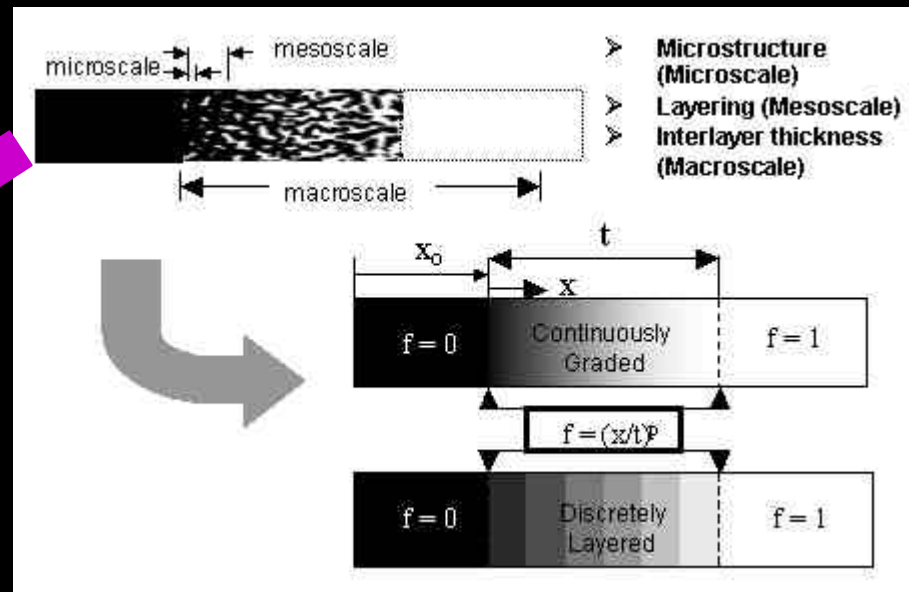
# Technical Objective



**Tailor Burn Rate Performance in a Monolithic Rocket Motor Utilizing New Design and Control Schemes for Twin Screw Extrusion based on Functionally Graded Material (FGM) Architectures**



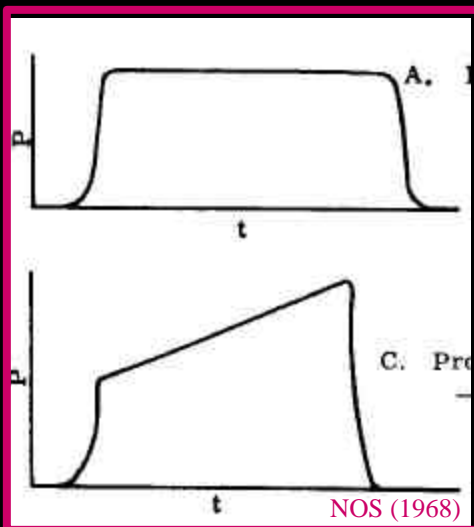
**Propellant Continuously Extruding from Die of TSE**



**FGM Architecture**



# Rocket Motors with Two Burning Rates



Examples of Conventional Rocket Motor Pressure-Time Traces

Functionally Graded Rocket Motor Pressure-Time Trace

$$P \propto \dot{r} S r = \dot{m}$$

$$\text{where, } \dot{r} = b \left( \frac{P}{1000} \right)^n$$

NOS (1968)

$$P \propto \dot{r}_x S r_x = \dot{m}$$

$$\text{where, } \dot{r}_x = b_x \left( \frac{P}{1000} \right)^{n_x}$$

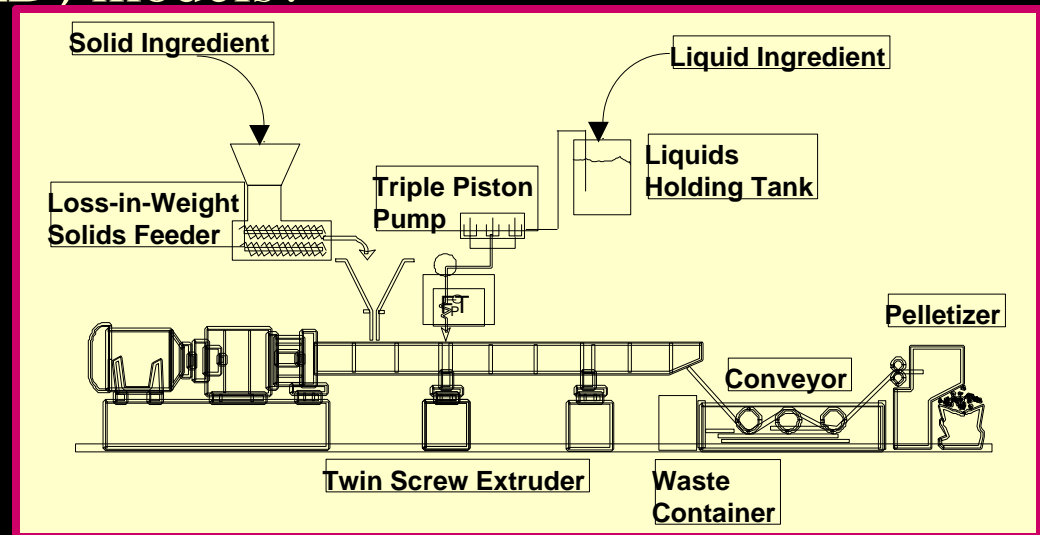




# Research Objectives



- How do dynamic variations in process conditions or ingredient addition during TSE affect the evolving microstructure of the extruded composite?
- Can the architectures be predicted by newly-developed residence distribution (RD) models?
- What characterization techniques have to be developed to adequately quantify microstructural variations in extruded material?
- How are these related to burning rate performance?



***Twin Screw Extrusion Process***



# Research Approach



**Manufacturing  
Science**

**Materials  
Characterization**

**Computational  
Tools**

*Inverse Design  
Procedure* – synergistic  
integration of component  
design with fabrication  
processes for optimizing  
performance using  
FGMs

Research is being con-  
ducted at UMD/College  
Park and NAVSEA-IH  
through a collaborative  
research agreement  
(*Center for Energetic  
Concepts Development*)

***Inverse Design  
Procedure***



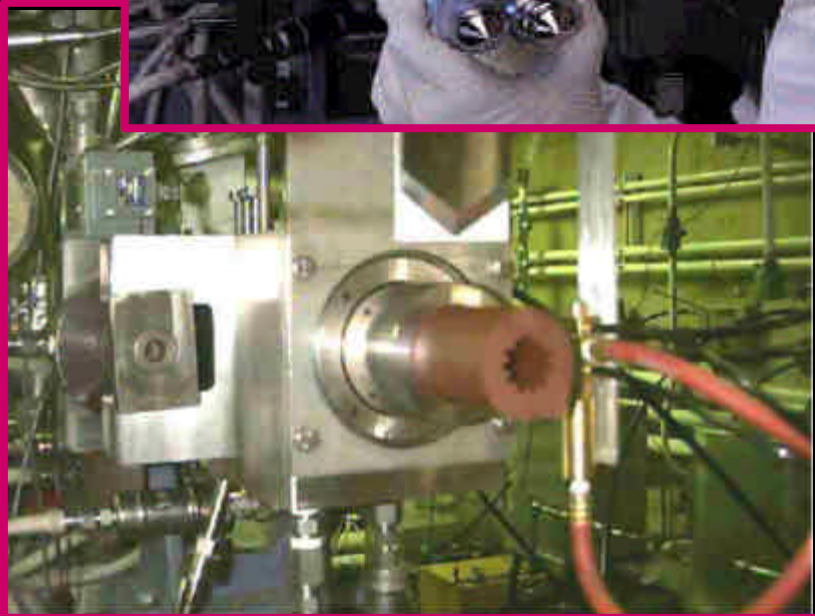
# *Advance State of the Art*



- Composite Energetic Materials have been traditionally manufactured using batch processing
- Current manufacturing of composite energetic materials is focused on homogeneous formulations
- New continuous manufacturing technology known as Twin Screw Extrusion (TSE) is being used to produce higher quality composite energetic materials with more flexibility and control
- The continuous nature of the TSE process is ideally suited for the manufacture of functionally graded materials (not restricted to energetic material)



# *40 mm TSE at Indian Head Process Technology Division*



- Pilot Scale, 15 - 100 #/hr
- World-class research facility for energetics
- New and existing propellants and explosives
- Refill capability
- Four LIW feeders, gear & triple-piston pumps
- On-line QA
- Ingredients to grains in one facility in one processing step

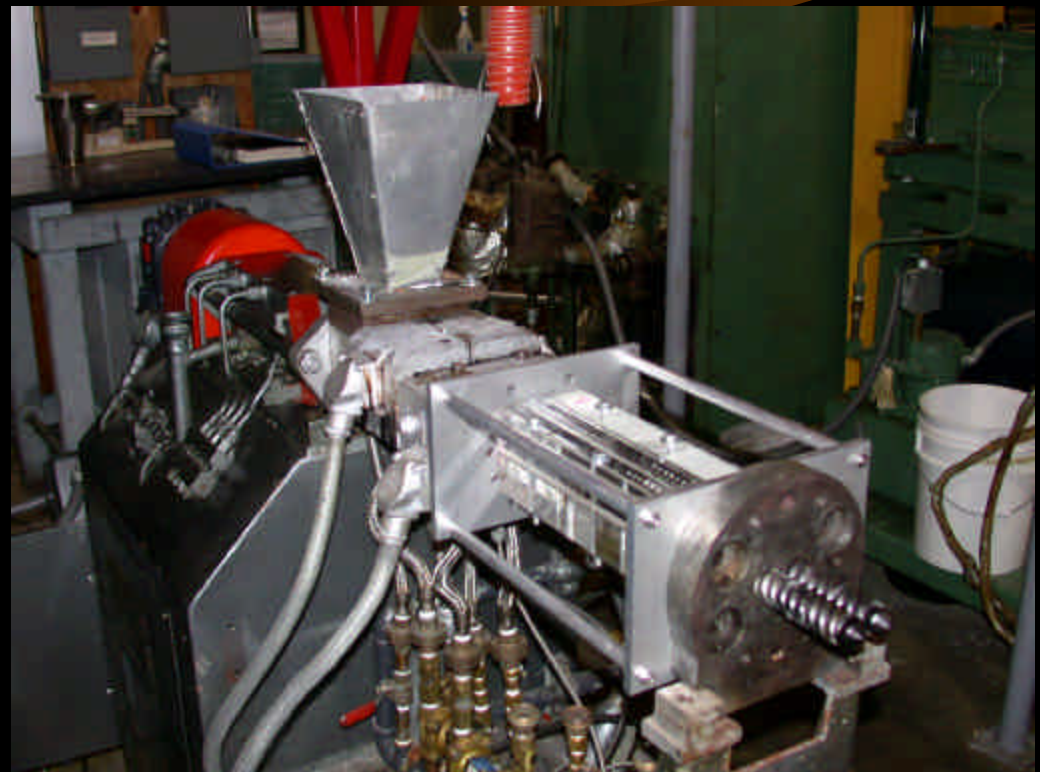




## *28 mm TSE at UMD*

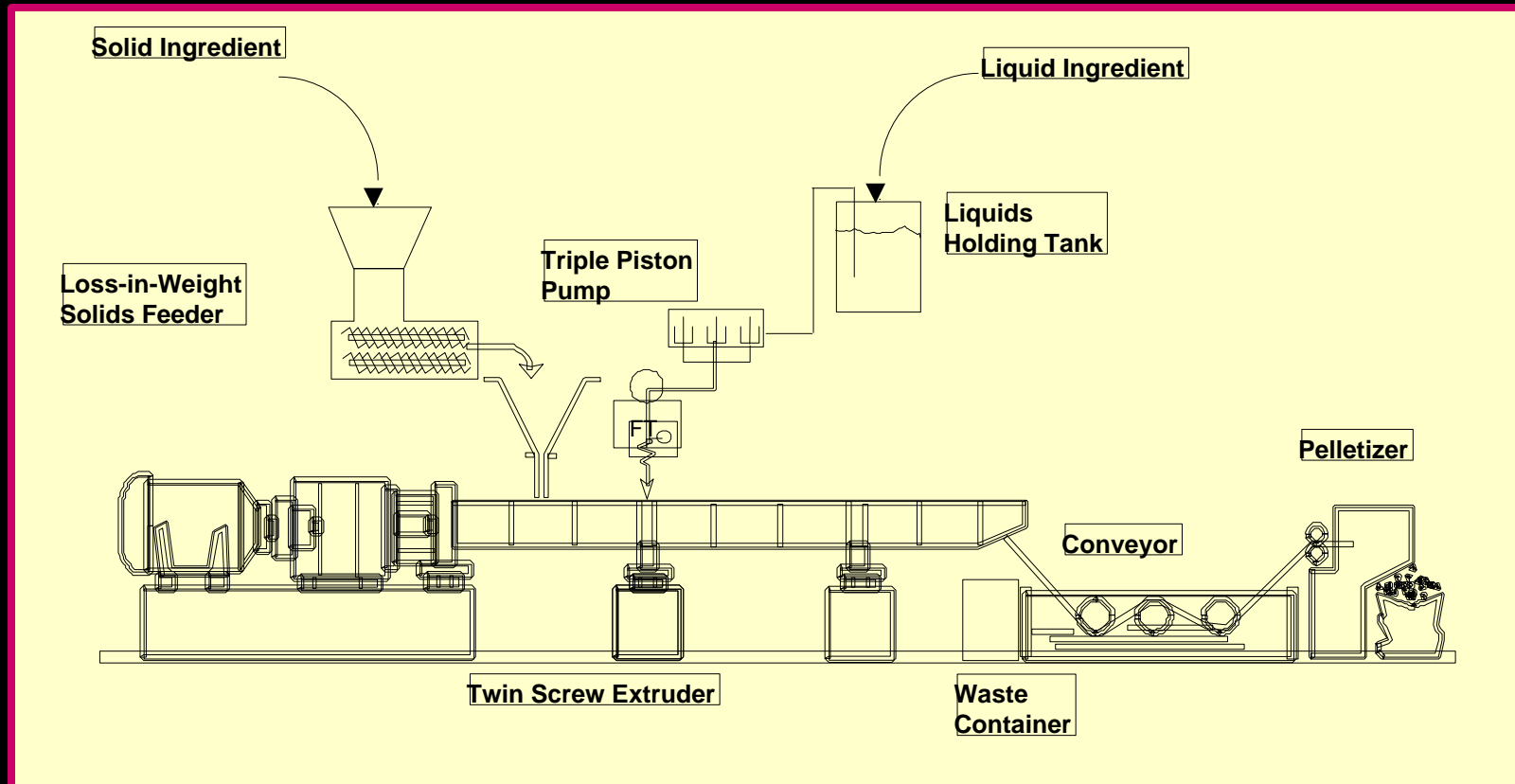


- Polymer Processing Laboratory - Dr. Bigio
- Corotating, fully intermeshing
- Research scale, < 10 #/hr
- Available torque is low
- Highly flexible process section
- Large screw inventory
- TSE Installed in 2001
- Feeders Upgraded to Loss-in-Weight Control 2002





# Continuous Processing using a Twin-Screw Extruder





# *Residence Distribution (RD) Modeling*



- Description of flow through stirred tank reactors (Danckwerts, 1953)
  - Age distributions
  - Distribution functions
  - Experimental determination with tracers
- Characteristic description of dampening due to backmixing (Rauwendaal, 1986)
- Characterize the ability of the process to dampen disturbances (Gao, Walsh *et al*, 1999)
- Experimental method: impulse addition of tracer
- Concentration of tracer at exit (or other location)
  - Function of time (Danckwerts)
  - Extended to Volume and RPM domains (Gasner, Bigio *et al*, 1999)



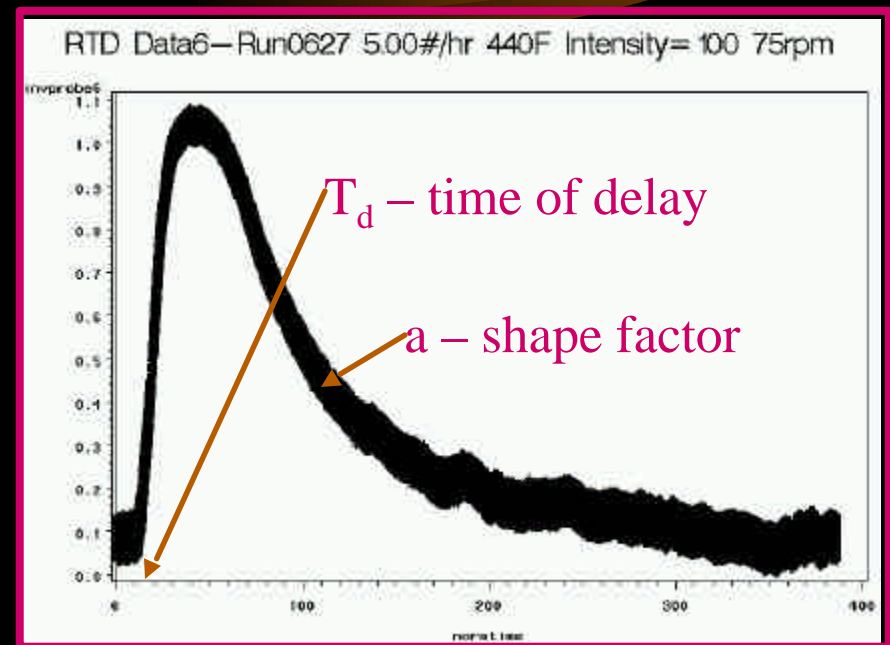




# Gao's Residence Time Distribution (RTD) Model



- Physically-based model
- Screw geometry
- Experimentally derived constant
- Quantifies the effect of a disturbance (e.g., ingredient change to make a FGM)



$$f(t) = \frac{a^3}{2} (t - t_d)^2 e^{-a(t-t_d)}$$



# Material Transport in TSE



Change in Filled Region Length with Respect to Time

$$\frac{dL_f}{dt} = \frac{(Q_{in} - Q_{out})}{HW(1 - \Phi)}$$

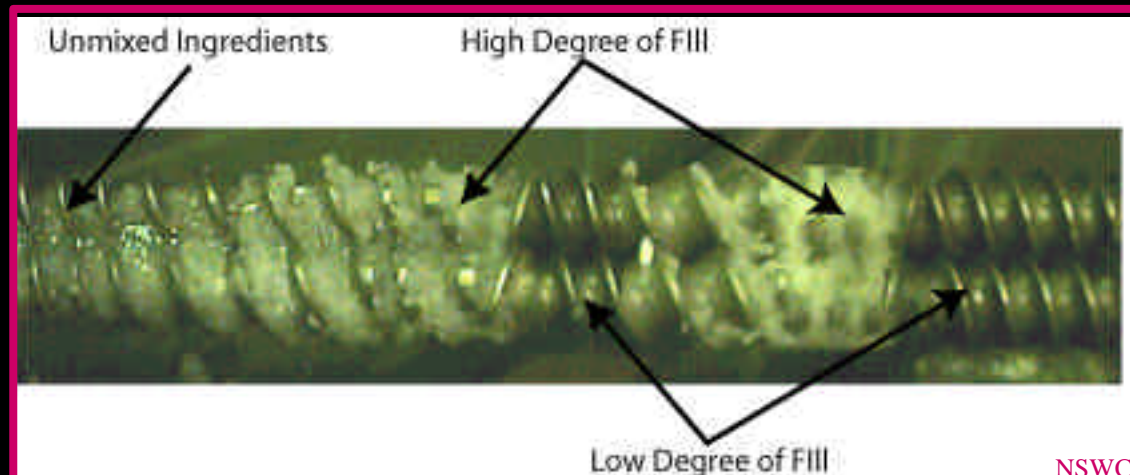
Flow into a Filled Region Described by Starved Flow

$$Q_{in} = Q_{st}(L_{st}, t) = Q_{feed} \left( t - \frac{2(L - L_f)}{V_{bz}} \right)$$

Flow out of a Filled Region Described by Die Flow

$$Q_{out} = Q_{die} \left[ \text{die geom., fluid props., } \left( \frac{dP}{dx} \right)_{die} \right]$$

Mudalamane (2002)





# Residence Volume Distribution (RVD)



Delay Volume ( $v_d$ )

$$v_d = t_d \times Q = A - \frac{3}{C} + B \frac{Q}{N}$$

$$g(v) = \frac{c \left( \frac{v}{Q} \right)}{\int_0^{\infty} c \left( \frac{v}{Q} \right) dv} = \frac{e \left( \frac{v}{Q} \right)}{Q} = \frac{a_v^3}{2} (v - v_d) e^{-a_v (v - v_d)}$$

RVD Curve  
Shape Parameter

$$a_v = \frac{a}{Q} = C$$

Residence Volume Distribution



# RVDs of Inert Composite Propellant using 40mm TSE



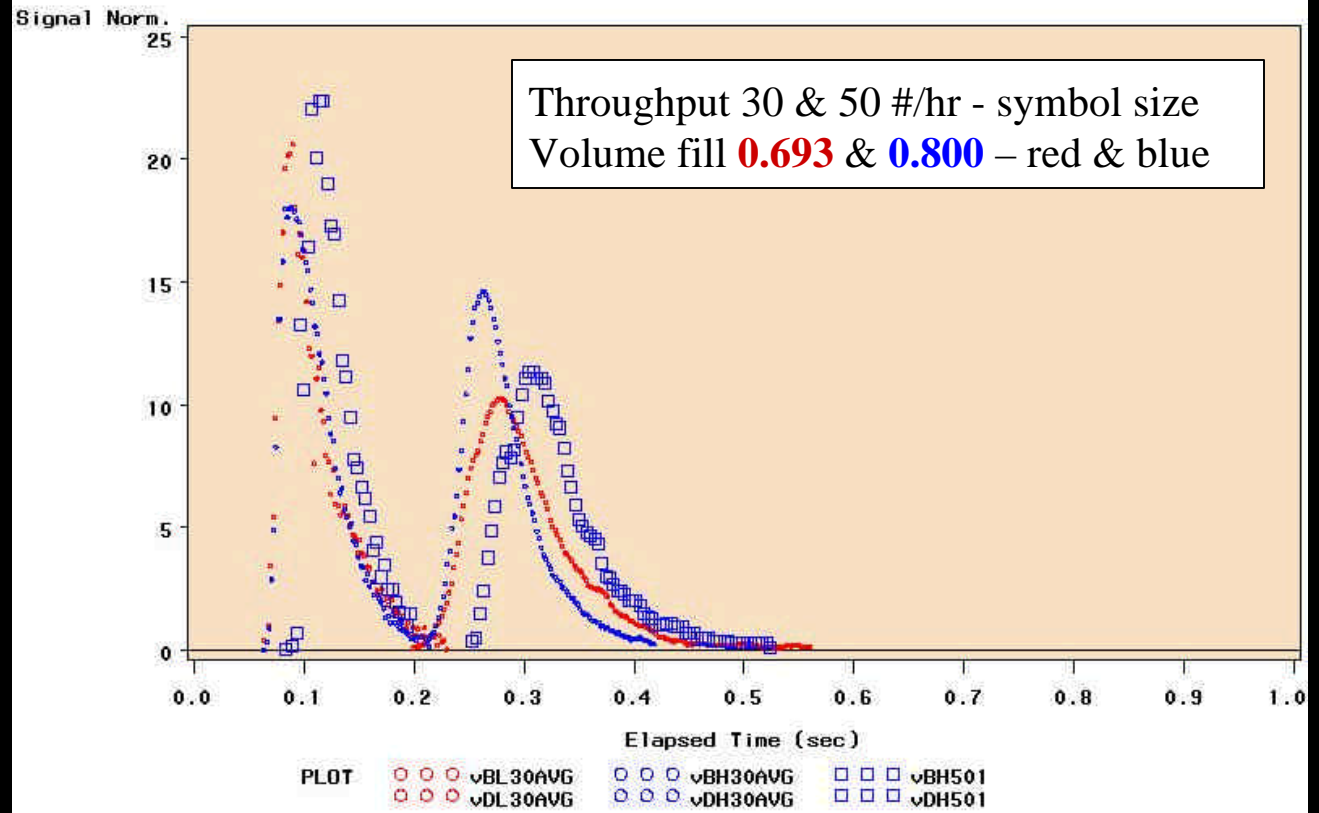
-RTDs for Inert Composite as Measured at Second Mixing Zone and the Diehead.

-RVDs not created yet.

-Similar measurements with live propellant conducted. (Data not analyzed yet.)

Plot of RVDs as a Function of phi and Q

FGM02i 30 & 50#/hr phi=0.693 & 0.800  
Sensor at Filler Mixing Zone and Diehead



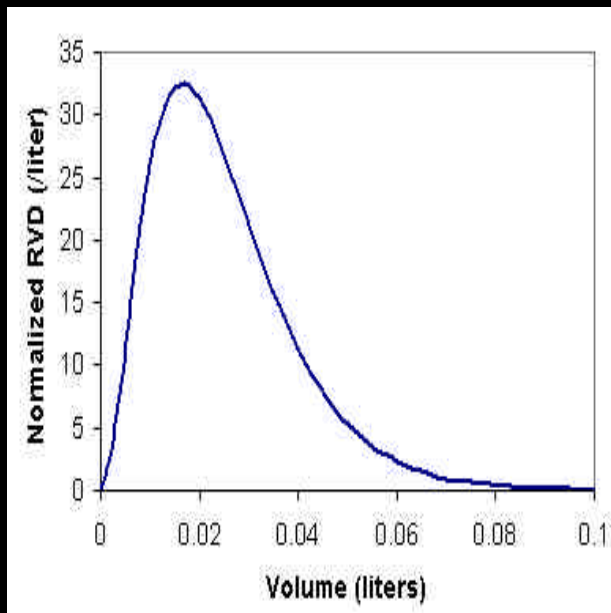




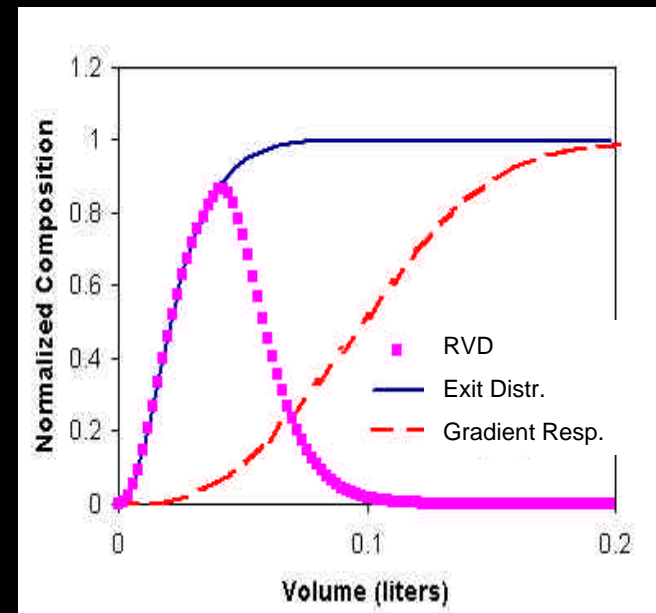
# Prediction of Gradient Architecture



Convolution of RVD to predict responses to step and ramp inputs



$$f[z(v)] = \int_0^v g(v-v')h(v')dv'$$

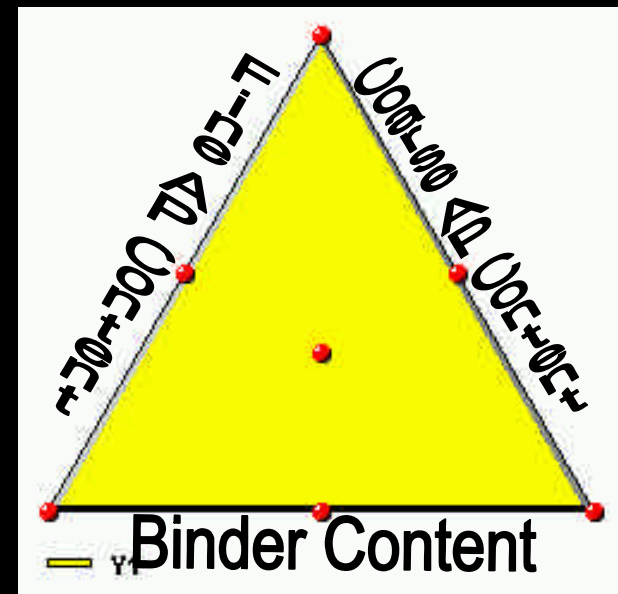




# Potential for Creating Gradient using IH-AC3



- Process Constraints
  - Two mixing stages process
  - Available solids feeders
  - Powder properties
- Safety Constraints
  - Ultra high fill
  - Too far from optimum ratio of coarse/fine
- Time Constraints
  - Limited manuf. budget
  - Multiple samples per extruder

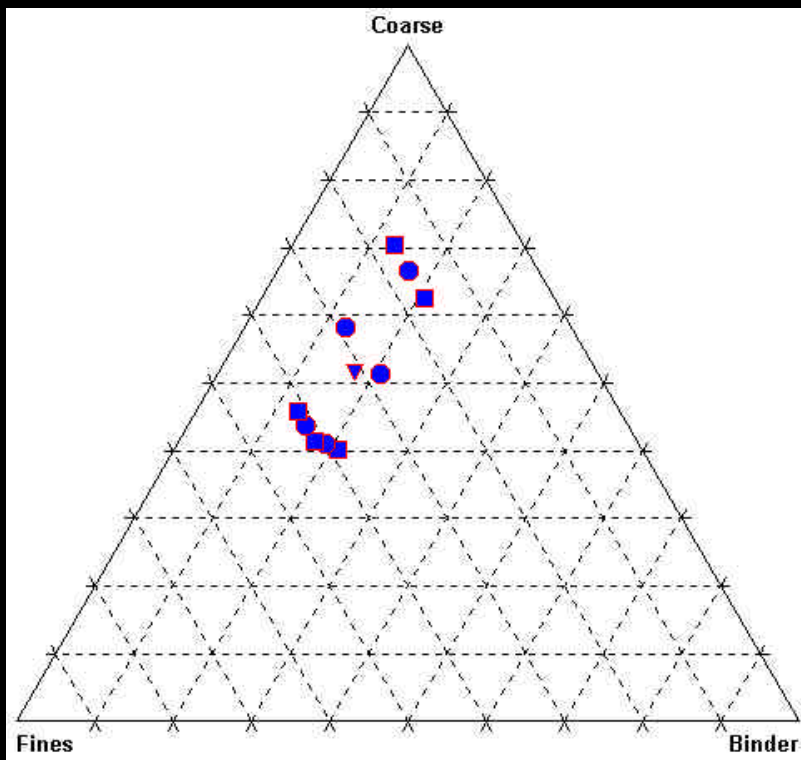




# Mixture Experiments



## Extreme Vertices Design



- Effects of individual ingredients on burning rate
- Three constituents for IH-AC3
- Combined effects of constituents
- Analysis by response surface methods
- Constrained region, Snee (1975)
- Cornell (1990)



# *Microstructural Characterization*



- Techniques Common to Metals, Ceramics, and Propellants
- *Functionally Graded Materials* are Highly Nonhomogeneous:
  - Structure is Designed
  - Structure Varies with Position
- Stereological Methods for Grain Size and Topological Distributions  
Liu (2000)
  - Scanning Electron Microscopy
  - Optical Stereoscope
- Transmission Electron Microscope, Energy Dispersive X-ray Spectroscopy
- Physical Properties
  - Stress-strain relationships
  - Micro-indentation



***Versamet Stereoscope and  
Optics System***





# Stereological Analysis



- Average Individual Particle Size

$$d_{av} = \frac{2}{p} \int_0^{p/2} \left( \frac{\cos^2 q}{d_{max}^2} + \frac{\sin^2 q}{d_{min}^2} \right)^{-1/2} dq$$

- Average Particle Size for the Distribution of Particles

$$\bar{d}_{av} = \sum g(d_{av}) d_{av}$$

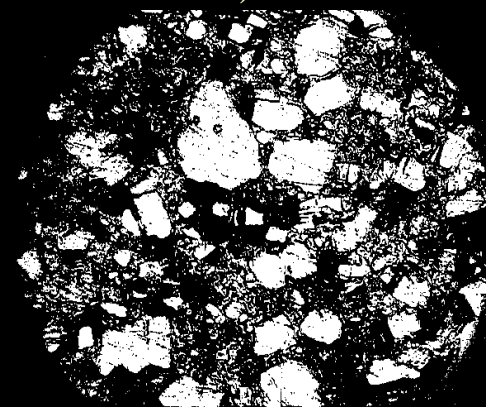
- Similar Treatment for Shape Factor Analysis

$$s_f = \frac{d_{min}}{d_{max}}$$

$$\bar{s}_f = \sum g(s_f) s_f$$

- Texture Analysis of Binary Images, Ohser (1998)

- Linear filtering
- Specific line length (volume fraction)
- Length of total projection (specific surface area)
- Integral of curvature (specific mean curvature)



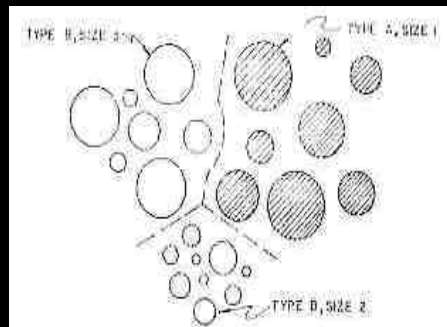
Binary Image of IH-AC3 Simulant



# Petite Ensemble Model



- Statistically-based combustion model
- Combines Beckstead, Derr, and Price (BDP) model with Glick's statistical formalism
- Models composite propellant as a random arrangement of polydispersed pseudopropellants

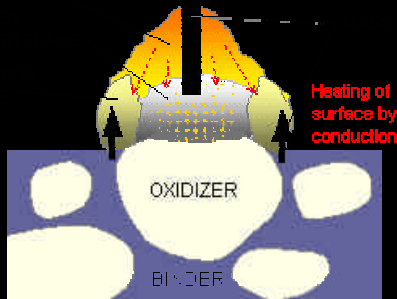


Polydispersed  
pseudopropellants

$$F_d = \frac{1}{(2p \ln s)^{1/2}} \exp \left[ -\frac{1}{2} \left( \frac{\ln D_o - \ln \bar{D}_o}{\ln s} \right)^2 \right]$$
$$\bar{r} = \int_{D_o} \frac{r_d F_d}{a_d} d(\ln D_o)$$
$$R_p = \frac{1}{\bar{r}} \int_{D_o} \frac{R_{p,d} r_d F_d}{a_d} d(\ln D_o)$$
$$R_v = \frac{1}{\bar{r}} \int_{D_o} \frac{R_{v,d} r_d F_d}{a_d} d(\ln D_o)$$



# Steady-State PEM Calculations (w/o fuel)



$$\bar{r} = m_{ox}^p / r_p$$

$$m_{ox}^p = m_{ox}^T \left[ 3 \left( \frac{h}{D_o} \right)_+^2 + 3 \left( \frac{h}{D_o} \right)_- + 1 \right]$$

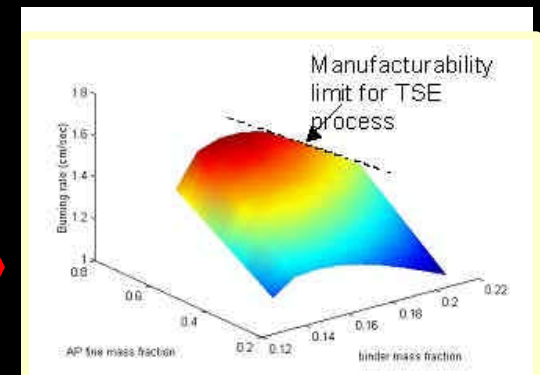
$$m_{ox}^T = A_{ox} \exp \left[ - \frac{E_{ox}}{RT_{s,ox}} \right]$$

$$\left( \frac{h}{D_o} \right)_\pm = f(r_{ox}, t_{ign}, D_o)$$

$$r_{ox} = m_{ox}^T / r_{ox}$$

$$t_{ign} = \frac{C_{ign} D_o^{d_D+1}}{p d_p}$$

Gradient effects?



Burn rate variation w/ composition

- Use COE to determine  $T_{s,ox}$  from adiabatic flame temperature calculations (PEP, NASA SP-273)





# Conclusions



- Described techniques to quantify dynamic variations in ingredient addition during TSE
- Discussed methods to relate process dynamics to the evolved microstructure of the extruded composite
- Shown how residence distribution (RD) models may be used to predict architectures
- Presented characterization techniques that should quantify microstructural variations in extruded material
- Researching to relate microstructural variations to burning rate performance