

A COMPARISON OF TECHNIQUES FOR MODELING XDT IN EXPLOSIVES AND PROPELLANTS

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2003 Insensitive Munitions & Energetic Materials Technology Symposium March 10-13, Orlando, FL

ACKNOWLEDGEMENTS

- NSWC Carderock Division
 - Dave Wilson
 - Tom Burton
 - Bill Hoffman
- Lawrence Livermore National Labs
 - Rich Couch
 - Jack Reaugh
- Sandia National Labs
 - Bill Erikson
 - Doug Drumheller
 - Mel Baer
 - Paul Taylor
 - Gene Hertel
- Lockheed Martin Missiles & Space
 - Erik Matheson
 - Ed Olsen
- Other
 - Eric Lundstrom; Lundstrom & Associates

OVERVIEW

- Background The SDT methodology
- Phenomenological & Physical aspects of XDT
- PERMS approach for modeling XDT
- CDAR approach for modeling XDT
- Final Thoughts

BACKGROUND (SDT)

- 'Shock to Detonation Transition'
- Basis for most hydrocode reactive modeling techniques.
- Structural response primarily Hydrostatic.
- Thermo-mechanical equilibrium.



BACKGROUND (SDT)

Hydrodynamic Conservation

Mass

$$\frac{d}{dt} \int_{b_t} \mathbf{r} dv = 0$$

Linear Momentum

$$\frac{d}{dt} \int_{\boldsymbol{b}_t} \boldsymbol{r} \, \boldsymbol{v} \, d\boldsymbol{v} - \int_{\partial \boldsymbol{b}_t} \boldsymbol{s}^T \, \hat{\boldsymbol{n}} \, d\boldsymbol{s} = \boldsymbol{0}$$

Energy

$$\frac{d}{dt} \int_{\boldsymbol{b}_t} \boldsymbol{r} \left(\frac{1}{2} \, \mathbf{v} \cdot \mathbf{v} + u \right) dv - \int_{\partial \boldsymbol{b}_t} \left(\mathbf{s}^T \, \mathbf{v} - \mathbf{q} \right) \cdot \hat{\mathbf{n}} \, ds - \int_{\boldsymbol{b}_t} \boldsymbol{r} \, z \, dv = 0$$

Chemical Kinetics

$$\mathbf{I}' = f(\mathbf{I}, p)$$

BACKGROUND (SDT)

Mixed-Phase Thermodynamics

Mixture Density

 $\boldsymbol{r} = \boldsymbol{j}_{s} \boldsymbol{r}_{s} + \boldsymbol{j}_{g} \boldsymbol{r}_{g}$

Mixture Internal Energy

 $u = \boldsymbol{g}_{s} u_{s} + \boldsymbol{g}_{g} u_{g}$

 $\mathbf{v} = \mathbf{v}_s = \mathbf{v}_g$

Thermal Equilibrium

$$T = T_s = T_g$$

Structural Equilibrium

$$p = p_s = p_g$$

Constitutive Response

Equation of State

Linear Elasticity

$$p = \underline{f(\mathbf{r})} + \underline{g(\mathbf{r}, u)} \qquad \mathbf{S}_{ij} = \frac{1}{2}$$

'cold curve' Thermal contribution

 $\boldsymbol{s}_{ij} = 2\boldsymbol{m}_{ij} + \boldsymbol{l}\,\boldsymbol{e}_{kk}\boldsymbol{d}_{ij}$

- Low Amplitude/Long Duration (LALD) mechanical loading.
 - Delayed detonative or sub-detonative response.
- Phase Disequilibria (transport effects vs hydro timescale)
- Plasticity $(J_1, J_2) \Rightarrow$ Thermal Dissipation.
- Damage (morphological change) ⇒ Sensitization.
 Example Low-speed bullet impact (bare explosive charge)



 Continuum Mixture Theory (Truesdell, 1969) Mass



Linear Momentum

Energy

$$\frac{d}{dt} \int_{\mathbf{b}_t} \mathbf{j}_a \mathbf{r}_a \mathbf{v}_a dv - \int_{\partial \mathbf{b}_t} \mathbf{s}_a \cdot \hat{\mathbf{n}} ds = \int_{\mathbf{b}_t} \mathbf{m}_a^+ dv \mathbf{v}$$
 Inter-phase drag, etc.

Conduction, Convection, etc

$$\frac{d}{dt} \int_{\boldsymbol{b}_t} \boldsymbol{j}_a \boldsymbol{r}_a \left(\frac{1}{2} \boldsymbol{v}_a \cdot \boldsymbol{v}_a + \boldsymbol{u}_a \right) dv - \int_{\partial \boldsymbol{b}_t} \left(\mathbf{s}_a^T \boldsymbol{v}_a - \mathbf{q}_a \right) \cdot \hat{\mathbf{n}} \, ds - \int_{\boldsymbol{b}_t} \boldsymbol{j}_a \boldsymbol{r}_a \boldsymbol{z}_a \, dv = \int_{\boldsymbol{b}_t} \boldsymbol{u}_a^+ dv$$

Dissipated plastic work

Total work

$$W = \int_{0}^{t} \boldsymbol{s}_{ij}(\boldsymbol{t}) \boldsymbol{e}_{ij}(\boldsymbol{t}) d\boldsymbol{t}$$

Strain Decomposition

$$\boldsymbol{e}_{ij}(\boldsymbol{t}) = \boldsymbol{e}_{ij}^{e}(\boldsymbol{t}) + \boldsymbol{e}_{ij}^{p}(\boldsymbol{t})$$

Cauchy stress Decomposition

$$\boldsymbol{s}_{ij}(t) = s_{ij}(t) - \frac{1}{3} \boldsymbol{s}_{kk}(t) \boldsymbol{d}_{ij}$$

Dissipated plastic work

Compaction/Distension energy

$$W = \int_{0}^{t} \left[\underline{s_{ij}(t)} \boldsymbol{e}_{ij}^{e}(t) + \underline{s_{ij}(t)} \boldsymbol{e}_{ij}^{p}(t) - \frac{1}{3} \underline{s_{kk}(t)} \boldsymbol{e}_{kk}^{e}(t) - \frac{1}{3} \underline{s_{kk}(t)} \boldsymbol{e}_{kk}^{p}(t) \right] dt$$

Elastic strain energy

Elastic compression energy

- Damage / Morphological change
 - Distension \Rightarrow hot spot formation (hole burning)
 - Grain break up \Rightarrow surface area increase (grain burning)
 - Crack evolution \Rightarrow thermal pathways (flame spread)

XDT versus SDT

SDT	XDT	
Initiation via Strong-Shock	Initiation via Low-Amplitude / Long Duration loading	
'Infinitely-fast' kinetics propagated by shock	'Slow \Rightarrow Fast' kinetics propagated (initially) thermo-mechanically	
Kinetics directly linked to hydrostatic (EOS) response	Kinetics linked to tensorial stress state and damage histories (CM)	
Solid / Gas phases in thermo- mechanical equilibrium	Equilibrium approximation questionable	

PERMS Model

- Propellant Response to Mechanical Stimuli
 LLNL (Maienschein, Reaugh, and Lee) 1998
- Motivation Propellant 'fallback' scenario.
- Hydrodynamic Conservation
 - Single Phase
- Chemical Kinetics
 - Dual Reaction progress variables (I&G form)

$$I_{1}' = G_{0}(1 - I_{1})^{s_{0}}(h - a_{0})^{y_{0}} + G_{1}(1 - I_{1})^{s_{1}}I_{1}^{q_{1}}r'(p) + \left(\frac{S}{V}\right)_{0}(1 - I_{1})^{s_{2}}r'(p)$$
$$I_{2}' = G_{2}(1 - I_{2})^{s_{2}}(RI_{1} - I_{2})^{q_{2}}p^{y_{2}}$$

PERMS Model

- Mixed-Phase Thermodynamics
 - Thermo-mechanical equilibrium
- Constitutive Response.

$$\boldsymbol{s}_{y} = \left[\boldsymbol{s}_{y}^{0} + \boldsymbol{a} \ln(\boldsymbol{e}_{eps}')\right] \left(1 + \boldsymbol{b}\boldsymbol{e}_{eps}\right)^{n}$$
$$\left(\frac{S}{V}\right)_{0} = A\left(\boldsymbol{e}_{eps} - \boldsymbol{e}_{0}\right) \left\langle \boldsymbol{e}' \right\rangle$$

Mises Viscoplastic Flow Rule

Sensitization factor (modifies primary I&G rate)

CDAR Model

- Coupled Damage And Distension
 - Lockheed, SNL (Matheson, Drumheller, and Baer) 1999
- Motivation Propellant / Explosive XDT.
- Hydrodynamic Conservation
 - Two Phase
- Chemical Kinetics

$$c_{s}^{+} = -\frac{\boldsymbol{r}_{s}(\boldsymbol{j}_{s} - \boldsymbol{j}_{s}^{0})}{\boldsymbol{t}_{H}} - \boldsymbol{r}_{s}\left\langle\frac{S}{V}\right\rangle a p_{g}^{n} H(\boldsymbol{T}_{i} - \boldsymbol{T}^{*})$$
$$\left\langle\frac{S}{V}\right\rangle = \frac{6\boldsymbol{j}_{s}}{d_{p}}\left(\frac{\boldsymbol{j}_{s}^{0}}{\boldsymbol{j}_{s}}\right)^{\frac{1}{3}} \left(\frac{1 - \boldsymbol{j}_{s}}{1 - \boldsymbol{j}_{s}^{0}}\right)^{\frac{2}{3}}$$

Pressure dependence w/ thermal induction

Sensitization factor (Solid volume fraction dependence)

CDAR Model

Mixed-phase thermodynamics

- Distinct pressures, temperatures, velocities between phases
- Constitutive Response

 $\boldsymbol{e}_{eff}^{\prime vp} = A \left(N_0 + M \boldsymbol{e}_{eff}^{vp} \right) exp \left(-\frac{\boldsymbol{s}^{Dr} + H \left(\boldsymbol{e}_{eff}^{vp} \right)^{n_2}}{S_{eff}} \right)^{n_l}$ $\boldsymbol{e}_d^{\prime p} = -D_{eff} \left(\frac{p_s}{|p_s|} \right) \left[S_1 \boldsymbol{I}_1 \left| 1 - \frac{\boldsymbol{e}_{eff}^d}{\boldsymbol{e}_{max}} \right| |\nabla \cdot \boldsymbol{v}_s| \right]$ $+ S_2 \boldsymbol{I}_2 \left(\frac{|\boldsymbol{b}_s|}{\boldsymbol{b}_s^0} \right)^{n_l} \right]$

 $\frac{\mathbf{j}'_s}{\mathbf{i}} = \frac{\mathbf{c}^+_s}{\mathbf{r}} - \mathbf{e}'_d$

Viscoelastic / Viscoplastic Model

Tensile Damage & Distension

Compaction / distension law

PERMS versus CDAR

Included Physics

Physics	PERMS	CDAR
Equilibrium Approach	Thermo-mechanical	Non-equilibrium
Inter-phase transport	No	Yes
Constitutive Approach	J2 (viscoplastic / isotropic hardening)	J2 (viscoelastic / viscoplastic)
Plastic Work	No	Step Function
Shear Damage Coupling	Yes	Yes
Tensile Damage Coupling	No	Yes
Flame Spread	No	No

Final Thoughts

PERMS Model

- Straightforward approach
- Few material coefficients
 - Material characterization test costs minimized
- May not include necessary physics for XDT modeling
 - J1, J2 plasticity
 - Dissipated plastic work
 - Volumetric damage process
 - Flame Spread
- CDAR Model
 - Highly (overly?) complex approach
 - Many (~130) material coefficients
 - Material characterization test costs quite high
 - Includes most, but not all, required physics
 - J1, J2 plasticity
 - Dissipated plastic work
 - Flame Spread