

# Weapon System Concepts for a Future Gunship

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# Next Generation Gunship (NGG) Analysis of Alternatives (AoA)

#### *The Gunship IRD Requires a Transformational Capability*

Target set:

Enemy troops in contact with friendlies

Capabilities:

- Situational Awareness
- Precise, responsive, focused weapons effects
- Persistence
- 24-hour operations
- Survivability

Supports two of OSD's operational goals:

"Denying sanctuary to enemies by providing persistent surveillance, tracking and rapid engagement with high-volume precision strike, ... against critical mobile and fixed targets at various ranges in all weather and terrains."

"Protect and sustain US forces in distant antiaccess and area-denial environments."



# AoA PSAS Enabling Technologies

- Hovering UAVs
- Common Operating Picture and Control (COPaC)
- Very Small Missile (VSM)
  - Precise, Responsive, Focused-effects Weapon
  - Prosecute multiple, simultaneous targets throughout the hemisphere under the aircraft
  - 4 inch diameter, 57 inch long; 45 pounds
  - Performance: 15 miles maximum, in 65 seconds
  - Guidance: GPS/INS (in-flight updates) + Laser seeker (optional terminal guidance)
  - Warheads: Lethal and Non-lethal
  - Cost: Approximately \$18,000 each



### Requirements

#### **Requirements**

- 10 to 15 nmi Range
- 360° Coverage
- Lethal
  - Enemy Personnel
  - Soft Skinned Moving and Stationary Vehicle
  - Armored Moving and Stationary Vehicle
  - Non-Hardened Structures
- Low Collateral Damage
  - Precise with Surgical Effects
- Deep Magazine = Light and Small
  - 45-50 pounds
  - 40-50 inches Long
- Affordable
  - Interservice Common Ammunition
  - Common System Components Across DoD/Commercial
- Selectable Effects
  - Modular Allowing for Guided Non-Lethal/Less than Lethal Munition Options
- Day/Night Capable
- Graceful Degradation
- Small Deployed Footprint
- Automated System



**Light Armored Vehicle** 



Buildings & Infrastructure



Personnel



Truck



Boat

#### VSM Capabilities are Best Provided by a Weapon System



# VSM Does Not Exist

-The Capabilities of the VSM are Not Performed by Munitions that Fit the Size and Weight Requirements of the VSM



# Mr. Wynne's Challenge

# Demonstrate Capabilities That Can Be Spiraled Into The Current Fleet





**Munition / Launcher Size** 

# 100mm < MUNITION DIAMETER <127mm

- Munition Volume Too Small for State of the Art Guidance & Propulsion Technology
- Lethal Payload Size Limited

- Munitions Too Heavy for one man lift
- Launch Loads Increase to Obtain Separation Velocities
- Lethality Overkill

#### 120mm AND 105mm DIAMETER MUNITIONS ONLY COMMONLY USED SIZES IN THIS RANGE



### **Concept Warhead**





#### 120mm vs. 4 inch Diameter Warhead Effectiveness Summary

Munition		Accuracy CEP	HOB (ft)	Number of Shots to achieve Pk		
				Target 1	Target 2	Target 3
Baseline	105mm (M1 MOD)	2 mils 2 mils	0 14	5 3	10+ 10+	10+ 10+
VSM	Fast (120mm)	1.5 m 5.0 m	3 4/5	1 1	1	2 3
	Slow (120mm)	1.5 m 5.0 m	3 4	1 1	1 1	1 2
	AoA VSM (4.0 in.)	1.5 m 5.0 m	3 4	1	2 2	3 7

• Effectiveness Studies Indicate that 120mm vs. 105mm/4 inch Diameter Warhead Designs Have Superior Performance Against Target Set



### Launcher

- Gun
- Missile
- Dispenser
- Things to Consider:
  - PSAS Must Stay Within Geographic Location of High Priority Mission
    - Orbit is Not Necessary for Gun Fire Control Allows for Short Range Use
      of Unguided Rounds
    - Less Expensive Munitions Can Be Used if Less Guidance Control Authority is Required (GPS Guided Artillery Rounds)
  - Gun Launch is a Well Proven Method of Deploying Munitions from Inside Aircraft
  - VSM-Like Munitions are Launched Out of Guns and Missile Launchers
  - Rocket Propelled Munitions Cannot Be Safely Ignited Inside of Aircraft
    - What if They Don't Leave?
  - Rocket Exhaust Plumes Burn Aluminum
  - Fully Maneuvering Rocket Propelled Munitions are Expensive Use Only When Needed
  - Unguided Munition are Cheap Use IF You Can

VSM must be developed in the context of an overall PSAS Combat System



#### 105-mm GUN vs. 120-mm MORTAR

#### 105-mm Howitzer

- Weight 100 rounds 4200 lbs
- Recoil Load: ~10,900 lbs
- Gun Recoiling Weight 1,465 lbs
- Muzzle Pressure: 3,560 psi
- Legacy System
- Little Guided Technology
  Ongoing
- Lethal Payload Size
- Current System Too Much Gun
- FCS 105mm Gun is Separate Loaded

#### **120-mm Breech Loaded Mortar**

- Weight 100 Rounds 3200 lbs
- Recoil Load: ~5,600 lbs
- Gun Weight 1,315 lbs
- Muzzle Pressure: 1,620 psi
- Leading FCS Fire Support Weapon
- Stryker Brigade Combat Team Fire Support Weapon
- A Lot of Guided Munition
  Development Work Ongoing
- Lethal Payload Size
- Low Gun Loads
- Lighter Ammunition

#### **120mm DIAMETER MUNITION OFFERS BEST OPTIONS**



- Fire Baseline 105mm M1 Howitzer Rounds and 120mm M931 Mortar Rounds from Modified M102 Howitzer
- Verify Functionality of 120mm Breech Loaded Mortar Concept, Establish Baseline Launcher for Unguided, Guided, and UAV Munitions
- Blast Overpressure Comparison
  - Reduce Blast Pressure on Aircraft Surfaces
- Recoil Force Comparison
  - Reduce Load on Aircraft
- 120mm Characterization
  - Range
  - Initial Flight Stability
  - Interior Ballistics





- Bore Barrel ID to 4.732 inches (120.2mm)
- Breech Ring Machine Larger Radius
- Breech Block Machine Larger Radius
- Extractors Machine Pockets To Fit 120mm Case













- Flight Stability of Mortar Round is Satisfactory
- Further Testing Recommended in Simulated Crosswind





- Purpose: Investigate in Detail the Performance of the VSM with 1100 fps Initial Launch Velocity
- 1100 fps found to be best velocity for weight, length of munition
- Look at Range Achieved for Various Flight Times Using Three Launch Methods:
  - Side
  - Top
  - Forward
- Expand Study to Include Lighter Munition (15 pound Warhead vs. 20 Pound Warhead) Capable of Meeting 15 nmi/65 second Requirement
- Investigate Low Level CONOPS and Ground Launch Range of Munitions



Configuration 2

Side Launched Initial Velocity = 1100 fps AOF = -90 deg



Downrange (nmi)







# AC-130 Spiral Launcher Approach

#### 120mm Breech Loaded Cannon Launching System

- 120mm Launcher/Munition interface
  - Lightweight Launcher and Ammunition
  - Lower Recoil Loads and Blast Overpressure
  - Safe Reliable Munition Separation
  - Multiple Service User Base
- Retains AC-130 Like Capabilities as Well as Embracing New Capabilities
  - Pylon Turn Not Necessary for Fire Control
  - CoPac Develops Instantaneously Computed Solution
- Multiuse
  - Conventional Unguided Ammunition
  - One or Two DOF Guidance as Well as Fully Maneuvering Munitions
  - Soft and Hard Launch of Munitions
- Automated Handling and Loading System
  - Tightly Integrated with CoPAC
  - Providing Health and Status of Ammunition and Launcher





**PSAS** Launcher

#### **PSAS Launcher Concepts:**

- Must address LO Platform Requirements
- Autoloaded Trainable Gun Launcher
- Fixed Tube Bank
  Launcher
- VSM works in either type











# **Munition Approach**

- Modular Open System Design
- Member of a family of munitions
- Lightweight
- Multiple Warhead Types
  - Blast Frag Penetrator
  - EFP
  - Combination Blast/Frag-EFP
  - High Pressure Low Blast Radius Warhead
  - Less-than-Lethal
- Remote Data Control Interface
- GPS/INS Guidance

#### Terminal Guidance Technologies Available

- Semi Active Laser Seeker
- GPS Only



Modular Munition Design and Standard Interfaces Allows Affordability



- 5 Shot Series from Airgun
- VSM Warhead Design
  - Inert
  - 4340 Steel Case
  - 20 pound
  - 120mm diameter
- 900 fps Impact Velocity
- Targets:
  - 10.5 5000 psi Reinforced Concrete; Normal Impact
  - 7 inch 5000 psi Reinforced Concrete; 30° Obliquity
  - .5 inch RHA; Normal Impact
  - .5 inch RHA; 30° Obliquity
  - 1 inch RHA; Normal Impact







Target Material	Target Thickness, in.	Obliquity, degrees	Impact Vel, ft/s	Complete Perforation	Exit Vel, ft/s	Projectile Assembly Weight, Ib
RHA Steel	1	0	892	No	n/a	22.30
RHA Steel	0.5	0	888	Yes	556	21.80
RHA Steel	0.5	30	896	Yes	655	22.01









Target Damage, Shot PGM-S03 .5" RHA 30° Obliquity

Penetration Hole in 0.5-in. RHA Steel



Target Material	Target Thickness, in.	Obliquity, degrees	Impact Vel, ft/s	Complete Perforation	Exit Vel, ft/s	Projectile Assembly Weight, Ib
Concrete	10.5	0	910	Yes	194	22.54
Concrete	7	45	914	Yes	242	22.27





Entrance Hole, Test PGM-C01 10.5" 5K RC Exit Hole, Test PGM-C02 7" 5K RC 30° obliquity





#### 7 inches 5000 psi Reinforced Concrete 45° Obliquity



# Summary

# **Very Small Munition:**

- 120mm Diameter
- Launchers:
  - -side firing gun capable of firing standard 120mm mortar ammunition
  - -PSAS Concept Launcher
- Provides 360° Coverage from either AC-130 or PSAS
- A Viable Spiral for AC-130
- Meets PSAS Roadmap Objectives











- M931 After Conversion
- Conversion Applicable to All M930 Series Ammunition





- Blast Overpressure of 120mm was Approximately 35% Lower than 105mm
- 120mm Mortar Rounds Shot at Highest Charge
  - Further Reduction in Blast Possible from Lower
    Propellant Charge Configuration



#### **Blast Pressure Curves**







- **Recoil Force of 120mm was** • Approximately 30% Lower than 105mm
- Recoil Mechanism Can be Optimized for **120mm Gun** 
  - Further Reduction in Recoil Force
  - Lighter Weight Gun Components
- Lower Charge Propellant Should Reduce • **Recoil Loads**







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- 120mm Muzzle Velocity -> 1218 ft/s
  - Maximum Muzzle Velocity from M120 Mortar -> 1040 ft/s
- 120mm Range: 10° QE -> 3400 m Average; 33° QE -> 7300 m Average
  - Maximum Predicted Range for 120mm -> 8100 m
  - Maximum Range from M120 Mortar ->7270 m



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- Design Pressure of Gun System Below Elastic Strength Pressure
  - Optimize Gun Dimensions to Reduce Weight and Footprint
  - Optimize Recoil Mechanism to Reduce Recoil Load







Plywood Backstop

Normal Impact RHA Steel Target Setup





#### 10.5 inches 5000 psi Reinforced Concrete 0° Obliquity





.5 inches RHA 30° Obliquity



# VSM Configuration Trade Studies Initial Velocity

- Purpose: Investigate Length and Weight Payoff of Utilizing Higher Gun Launch Initial Velocity vs. Carrying More Rocket Propellant in Munition.
- Fixed Performance: Munition to Achieve 15 nmi in 65 seconds
- Allow Necessary Weight and Length to Vary Given Different Initial Velocities
- Considerations:
  - G Load on Munition
  - Recoil Loads
  - Chamber Pressure
  - Munition Length
  - Munition Weight



## VSM Configuration Trade Studies Initial Velocity





# VSM Configuration Trade Studies Initial Velocity

- Higher Initial Velocity Allows Munition to Loose Weight and Length for Same Given Range Performance
- Limiting Factors
  - Chamber Pressure
  - G Loads on Munition 7500 G is not Difficult for Mortar Ammunition Components
- 1100 fps Yields Acceptable Levels for All These Parameters:
  - 51-pound Munition
  - 45 inches Long
  - 21,000-psi Chamber Pressure
  - 7200 G
- Lower Mortar Propellant Increments will Permit Lower G Launch, and Munition Weight and Length Benefits will be Achieved, Needs Further Investigation



- Side Launching Munitions Offers Some Performance Advantages Over Top and Forward Launch
- Top Launch has Best Extended Range Advantage, but has Some Difficulty Getting Under AC
- New Control Algorithms Being Examined for Better Offside Performance from Side Launched Munitions
- Low Altitude CONOPS are Feasible with this Munition
- VSM would have Tremendous Application as a Ground Combat Fire Support Munition
- The Range and Response Time Requirements must be Examined in the Context of the Overall PSAS Combat System









Top Launched Initial Velocity = 1100 fps AOF = -90 deg



Downrange (nmi)



#### **Configuration 1**

Forward Launched Initial Velocity = 1100 fps AOF = -90 deg





#### VSM Configuration Trade Studies Range Studies Top Launched

Top Launched Initial Velocity = 1100 fps AOF = -90 deg

#### Configuration 2



Downrange (nmi)



Forward Launched





- Purpose of Study: Develop a Simulation that will Allow Modeling of Munition Flyout from AC
- Six Degree of Freedom (6-DOF) Simulation
- Ejection Speed (Study Variable)
- Parameters:
  - C-130 Slip Stream and Prop Wash is Modeled
  - Munition is Unguided Through Separation
  - Munition Initial Conditions Based Upon:
  - Aircraft Velocity (240 kts)
  - Launcher Position Relative to Aircraft (X', Y', Z')
  - Two Launcher Angles Relative to Aircraft: Azimuth Ψp, Elevation θp
  - One Munition Angle Relative to Launcher: Azimuth Ψm





Global (Aircraft):X, YLocal (Launcher):X', YLauncher Azimuth:ΨρLauncher Elevation:θρMunition Azimuth:Ψm

X, Y, Z X', Y', Z' Ψρ θρ Ψm





- Initial Velocity: 100 fps
- Initial Velocity: 250 fps.
- Initial Velocity: 30 fps Good Trajectory
- Initial Velocity: 30 fps Bad trajectory



- Initial Studies of Munitions Indicates that Lower Launch Velocities (200-500) will not have Difficulty Separating from AC
- Oscillation of Munition During Low Velocity Launch May be an Issue for Guidance System
- Further Development of Simulation Warranted Using AFSOC Owned AC-130 Flow Field Model Developed by Auburn University
- Results Should be Verified in Actual Flow Field



#### VSM Configuration Trade Studies Motor Configuration

- Study of Pintle Motor Configuration Examined to Investigate Potential of this Type of Motor
- Possible Pintle Motor Trade-offs:
  - Fly During "Cruise" with Low Mass Flow Rate Allowing for Possible Higher Terminal Velocities
    - This Allows for More Propellant to be Left to Throttle Up With
  - Fly During "Cruise" with High Mass Flow Rate Allowing for Possible Shorter Time-of-Flights
    - This Leaves Little Propellant for Throttling Up, but can Possibly Give Better Time-of-Flights than a Boost/Sustain Motor
- This is only a rough cut! This pintle motor design will need some optimization to determine what the best burn conditions are for the specified mission (i.e. higher terminal velocities and shorter time-of-flights may be possible to achieve).



#### VSM Configuration Trade Studies Motor Configuration

Pintle vs. Boost/Sustain Motor 15 nmi Case Initial Velocity = 1100 fps





#### VSM Configuration Trade Studies Motor Configuration

- Rough Cut Analysis Indicates that the Configuration is Promising
- May Need to Be Soft Launched
- Less Mature Technology
- This Pintle Motor Design will Need Some Optimization to Determine What the Best Burn Conditions are for the Specified Mission
- With this Technology it may be Possible to Achieve:
  - Higher Terminal Velocities
  - Shorter Time-of-Flights