

Thermally Activated Venting System for IM

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Shape Memory Polymers for Insensitive Munitions

Phase I SBIR effort funded by MDA

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Technical Monitor Heather Simko – MDA/QS (Safety, Quality, & Mission Assurance)

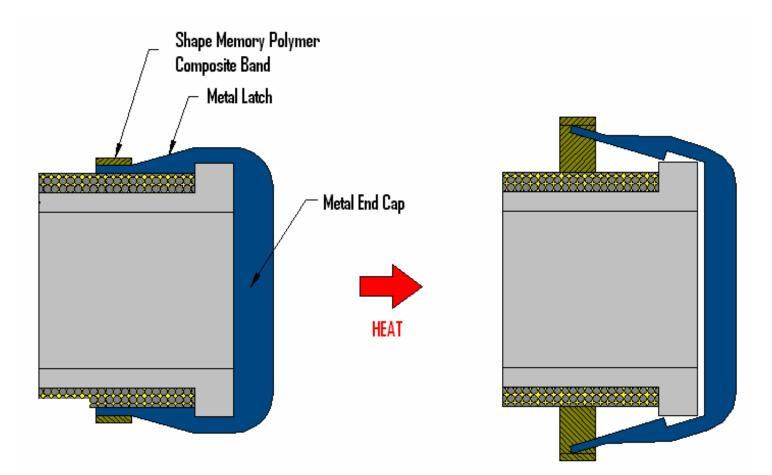
Phase II awarded

Initiated 2-year effort in June 2007



Original IM Technology Concept

Initial Potential Pressure-release Mechanism Design Using Shape Memory Polymer





Phase I Program Goal

Phase I Goal

Demonstrate the feasibility of shape memory polymer actuated venting systems for IM compliant solid rocket motors (SRM)

Phase I Achievements

SMP material compatible with propellant temperatures identified Designs for SMP venting mechanism options developed SMP prototype demonstrated



Determine Design Requirements

Temperature range depending on propellant type

Mechanical requirements

Environmental exposures

Temperature & Humidity Vibration (Transportation Requirements) Aging and Creep



Develop Design Requirement Process

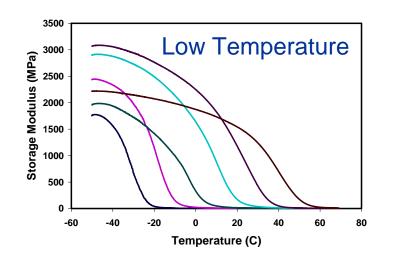
General 3 Step Design Process

Step in Design Process	Information Needed	Resulting Design Requirement
1. Dimensional design (CRG)	SRM dimensional envelope	Dimensions of SMP mechanism
2. SMP Material selection (CRG)	Propellant type, autoignition temperature	Transition temperature of SMP material
3. Assessment of SRM design impacts (SRM Manufacturer)	Transition temperature of shape memory polymer	Insulator/closure redesign



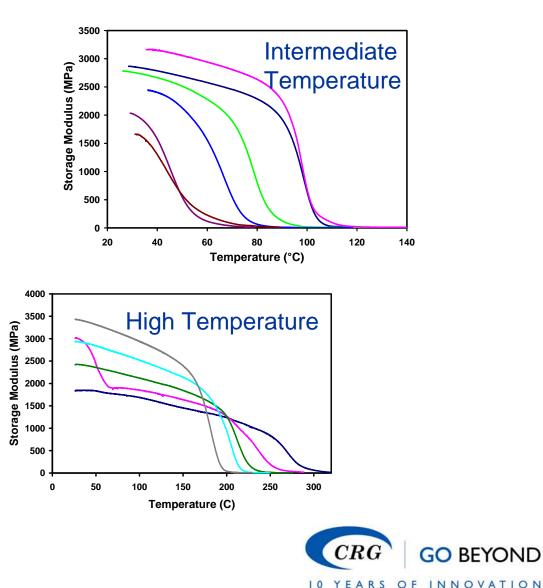
Identify & Select SMP

CRG can modify SMP materials to have a broad range of transition temperatures depending on the propellant type



Design principle applicable to most classes of polymers

- Acrylate
- Styrene
- Ероху
- Cyanate Ester
- Maleimide
- Etc...



Design prototype mechanism that uses SMP's unique properties for actuation

Several preliminary concepts: Initial composite band concept Reconfigurable thread concept C-clamp concept Turn and lock concept Internal retaining ring concept Plug concept External retaining ring concept











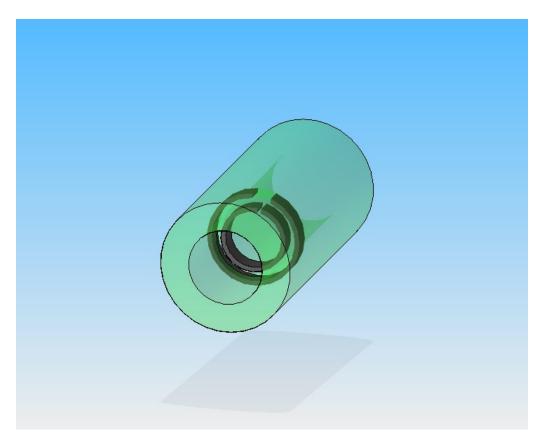
CRG Selection Criteria:

Limitation of load seen by shape memory polymer Significant amount of venting provided by mechanism activation Internal location of mechanism (external volume limited) Ease of integration into current and future systems

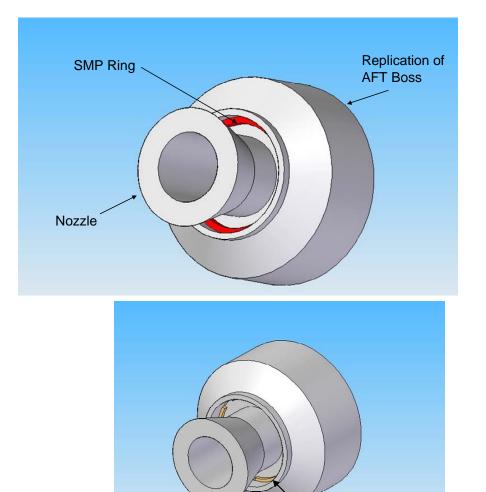


Concept Downselection

Internal retaining ring concept

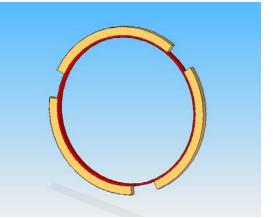






3 piece metal ring

Designed internal retaining ring based on a generic solid rocket motor concept



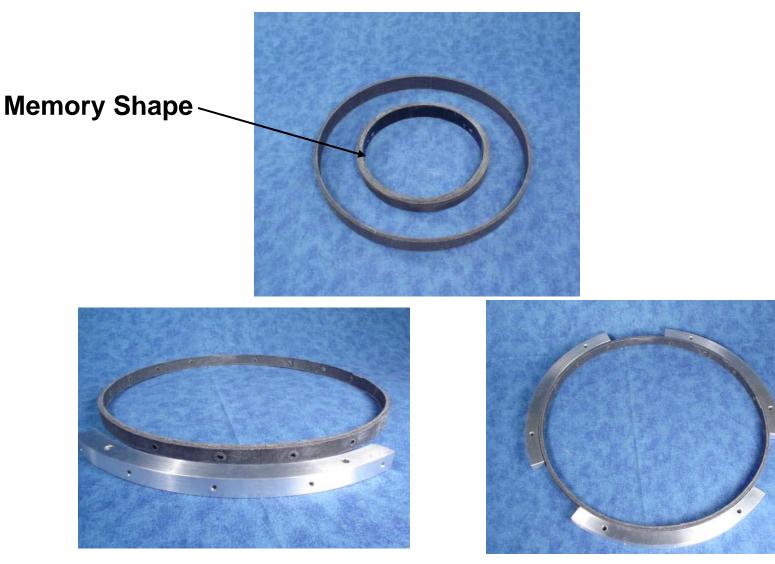
3 piece metal ring (yellow) with SMP snap ring (red)



Fabricated prototypes of selected design Used CRG's styrene-based resin system for demonstrator Basic concept demonstrator for analysis







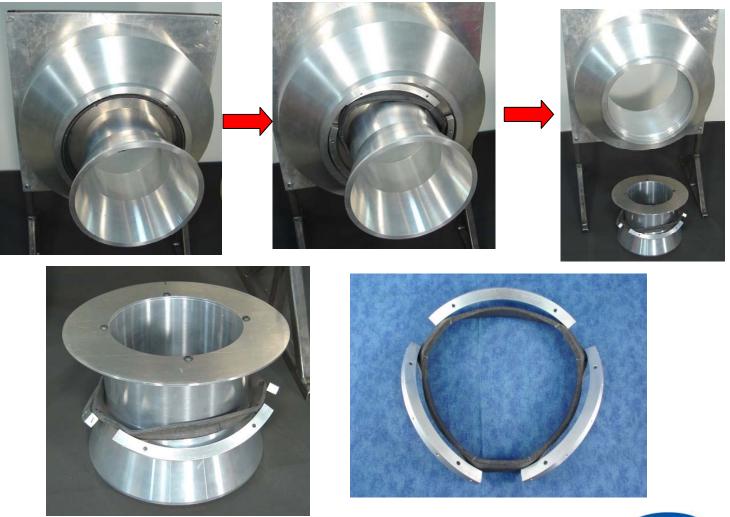
Expanded SMP and attached 3-piece metal ring







Prototype Demonstrator





Summary

Efforts focused on demonstrating feasibility

- **Assessed requirements**
- Developed general device mechanism concepts
- **Evaluated general concepts**
- Selected appropriate SMP materials
- Analyzed motor designs to
 - Identify and analyze potential integration opportunities
 - Downselect to a single concept for prototyping

Developed specific device design

Fabricated device prototype

Characterized prototype and assessed feasibility



SMP Mechanism Benefits

Enables venting for slow cook-off situations Fail-safe (non-electronic thermal venting system) Ability to tailor for varying propellants Simple designs available Easily adapted to many different solid rocket motor systems Can also be integrated into ammunition containers



Future Work – Phase II

Integrate SMP mechanism into an SRM for SCO and pressure testing Optimize design for a current or future MDA platform integration Conduct further materials characterization Look at other applications such as munition containers Develop system integration documentation Develop manufacturing plan

