



# **ECL<sup>®</sup> Propellant Demonstration Consolidation of 105 mm Artillery M67/M200 into Single Charge System Benefits for the Warfigther**

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“Approved for Public Release; Distribution Unlimited”

# Main Goals of Improvement Program

- Replace current charge system M67 (7 zones) and M200 (standalone long range)
- Create compact charge system with 5 – 6 zones with sufficient overlap capacity
- Facilitate handling for operation, improve reliability and shelf life for operation in complex terrain under extreme loads
- Optimize life cycle cost

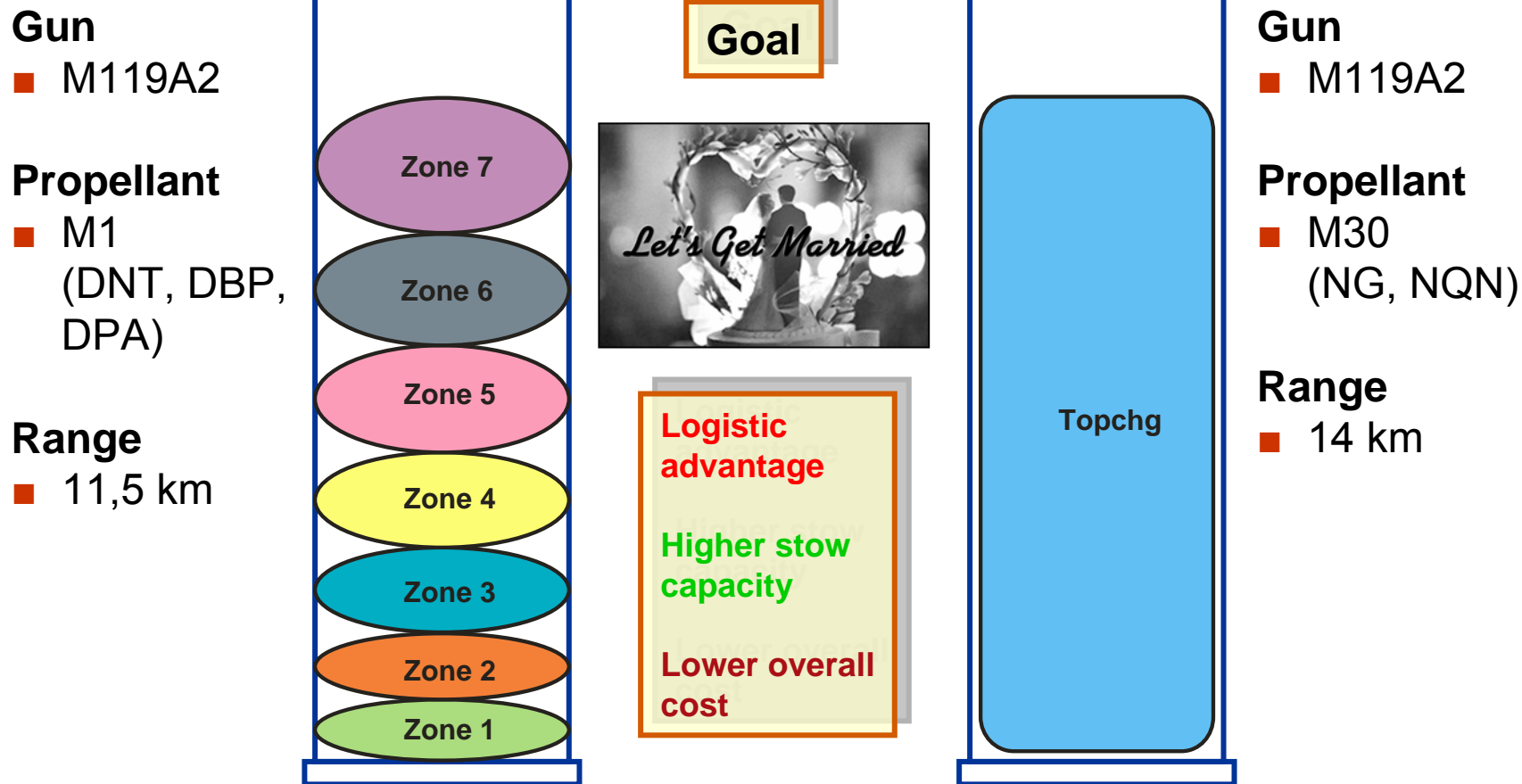
U.S. Army photo/1st Lt. Jonathan J. Springer



U.S. Army photo/1st Lt. Jonathan J. Springer



# Current Situation – M67 and M200



# Single Charge System with 6 Zones

## M 67

### Gun

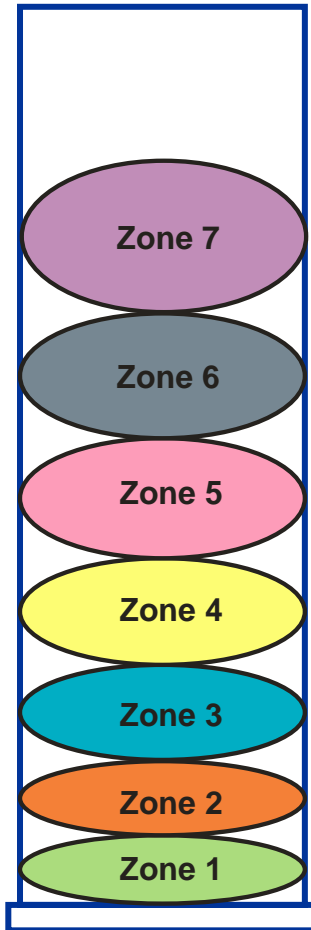
■ M119A2

### Propellant

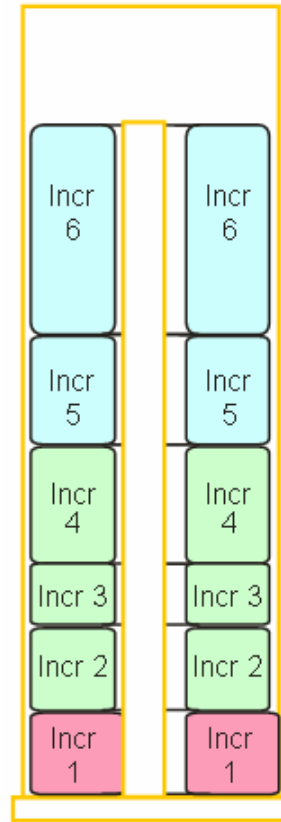
■ M1  
(DNT, DBP,  
DPA)

### Range

■ 11,5 km



## Conception



## M 200

### Gun

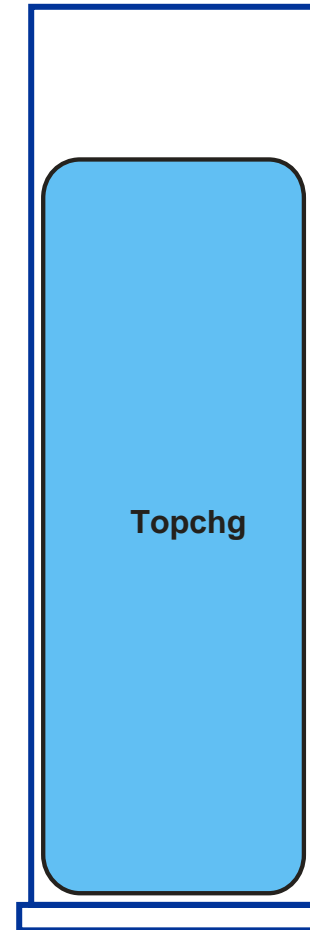
■ M119A2

### Propellant

■ M30  
(NG, NQN)

### Range

■ 14 km



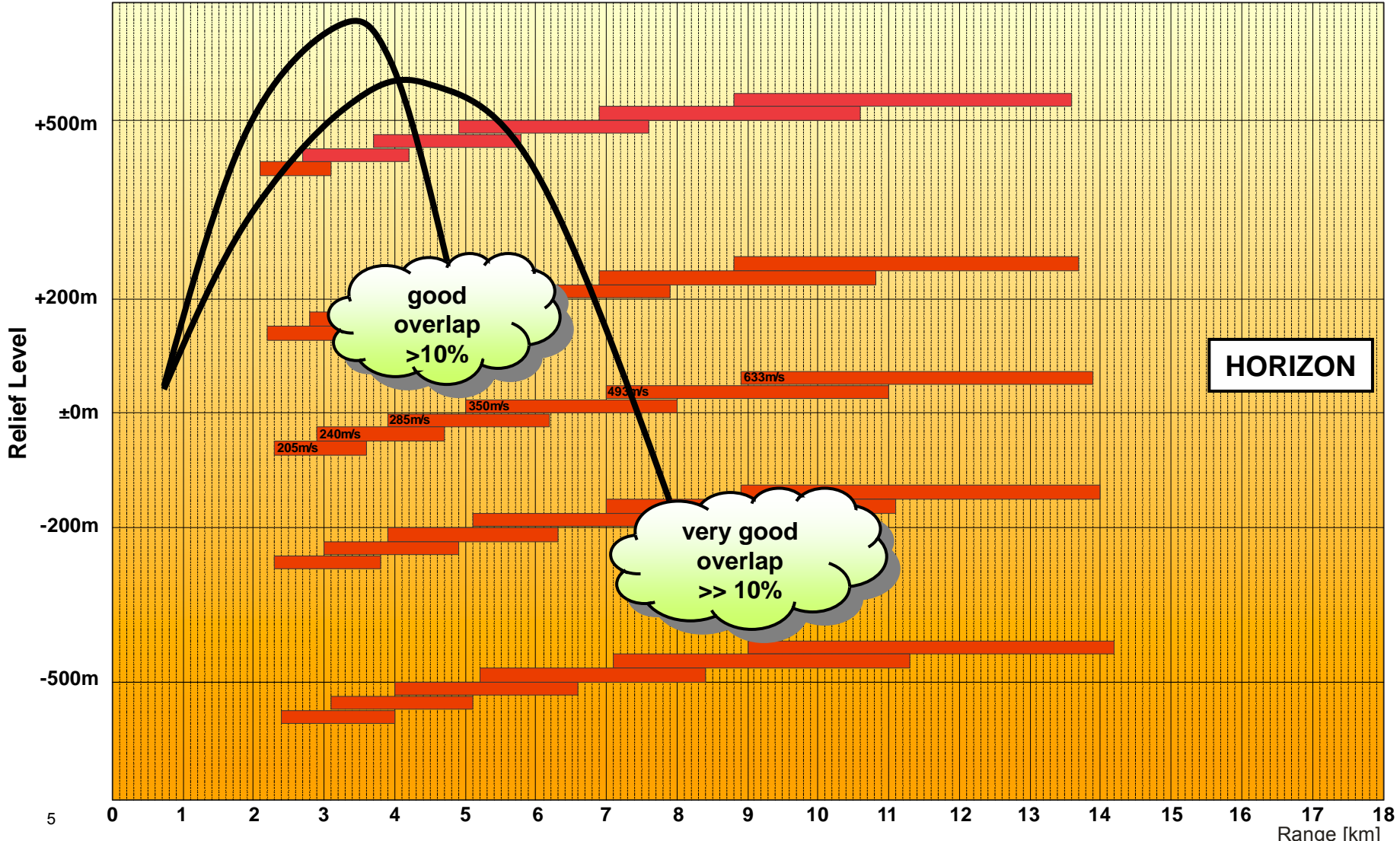
# Nitrochemie's Proposed 6 Zone Design Solution



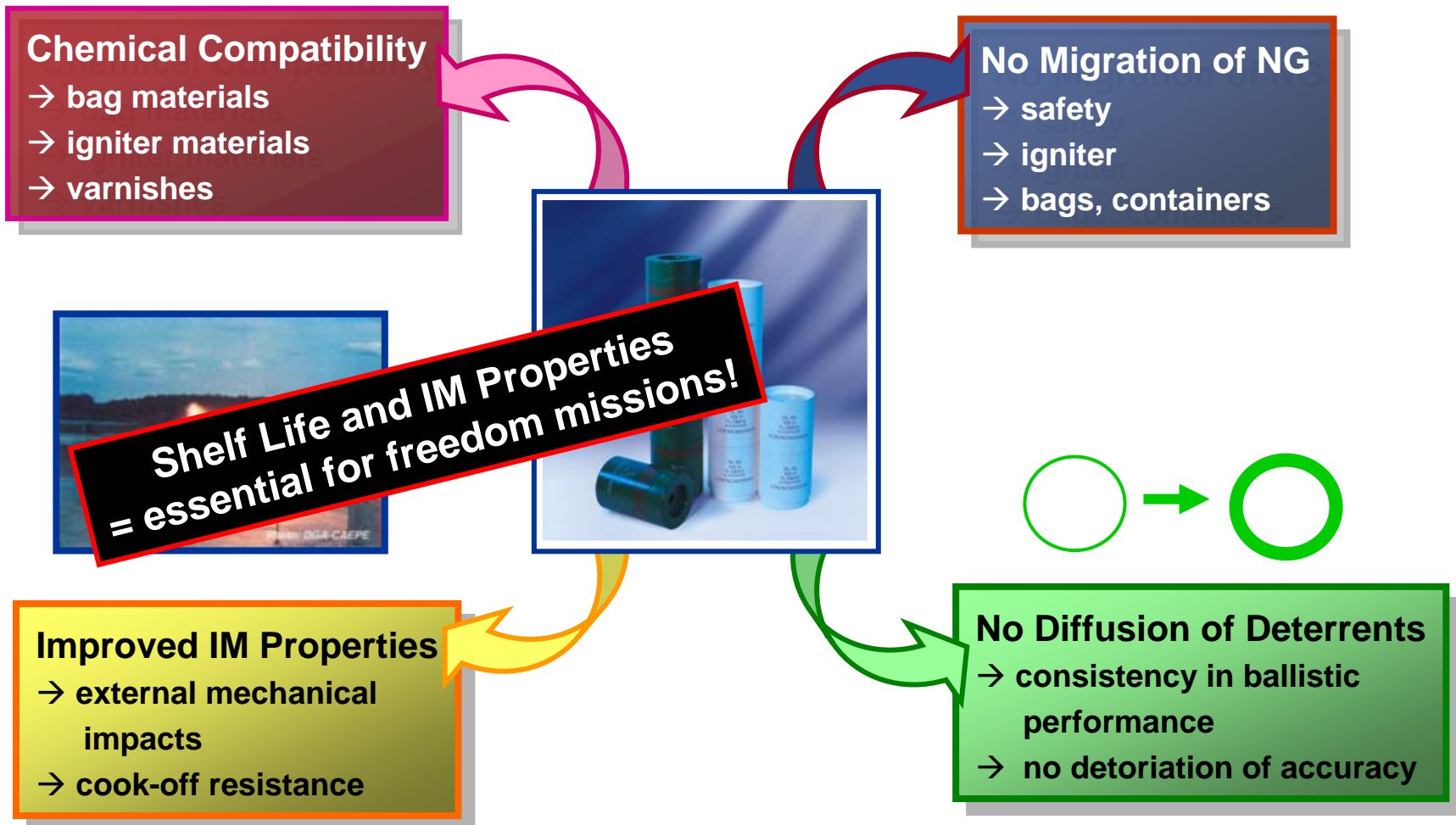
## General overlap situation in complex Terrain

Range min  
El. 1244A‰

Range max  
El. 800A‰



## General Benefits over *NG containing Propellants*



## Donut Bags Improve Loading and Handling of Propellant Charges



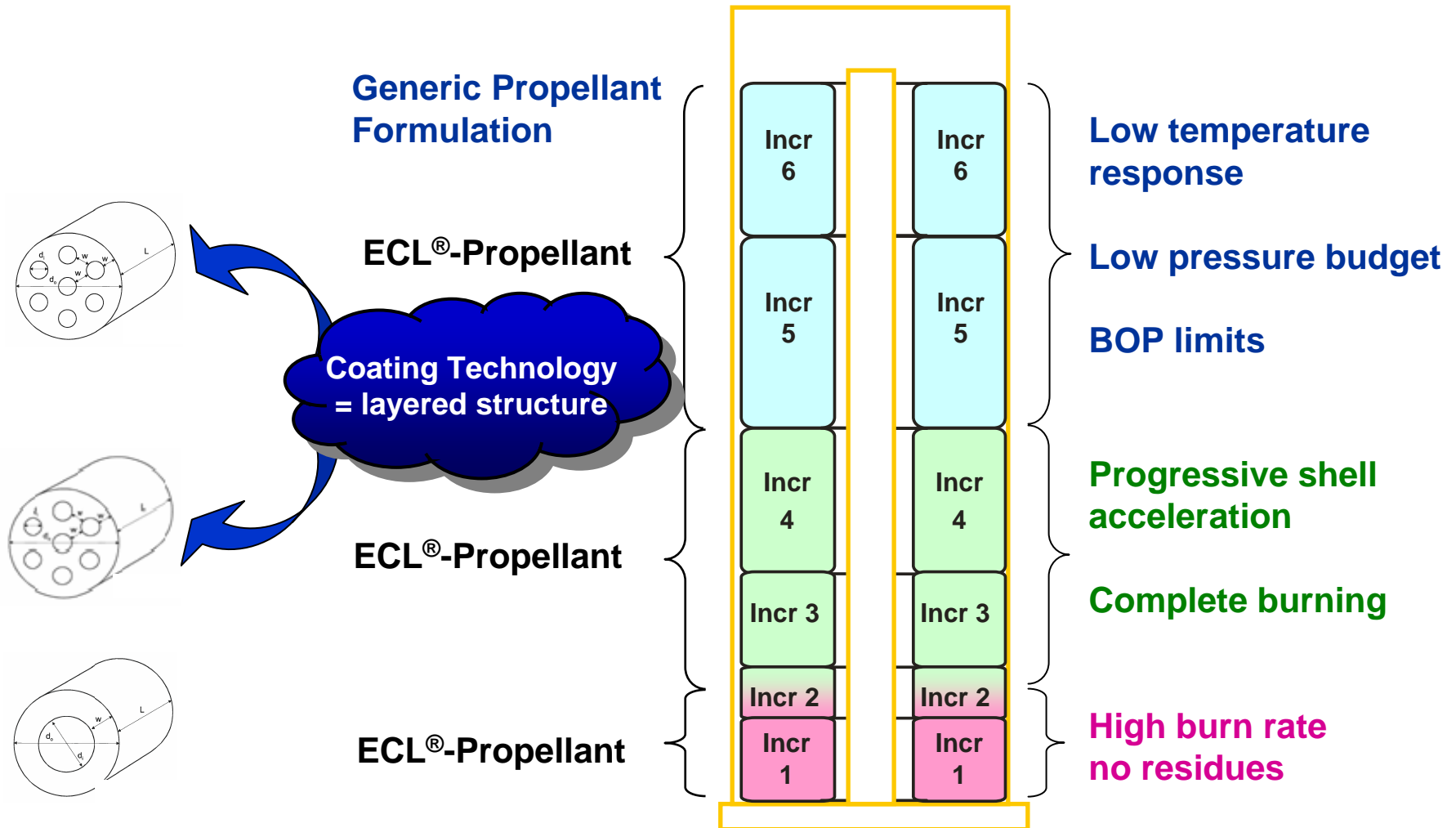
Current bag design (M67 and M200)

➔ poor loading capacity of M67

Proposed design for single integrated charge system (XM350) includes donut-style bags of 3 different sizes providing:

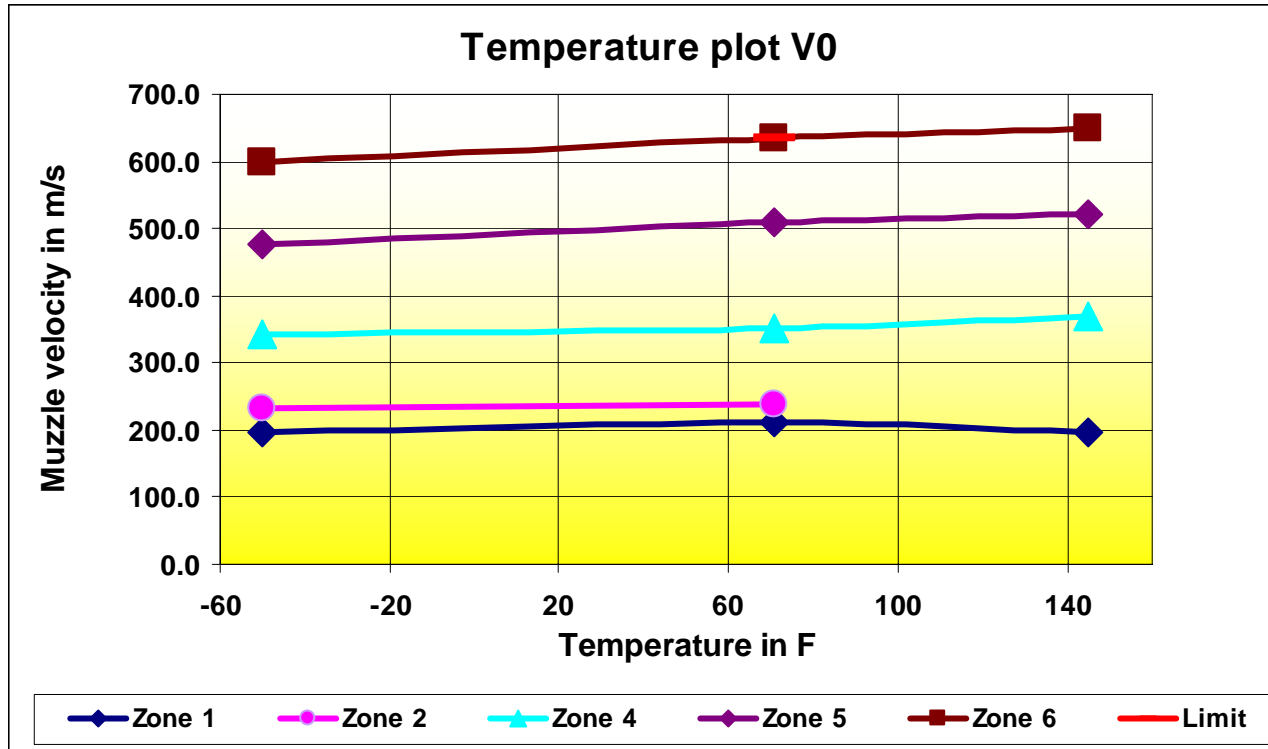
- **optimal loading capacity!**
- **easy handling for troops**

## Tailored ECL<sup>®</sup> Propellant Design for 6 Zone Concept





## Results of Velocity Measurements



### Goals @ 70F

Zone 1 199m/s

Zone 2 238m/s

Zone 3 285m/s

Zone 4 350m/s

Zone 5 501m/s

Zone 6 652m/s

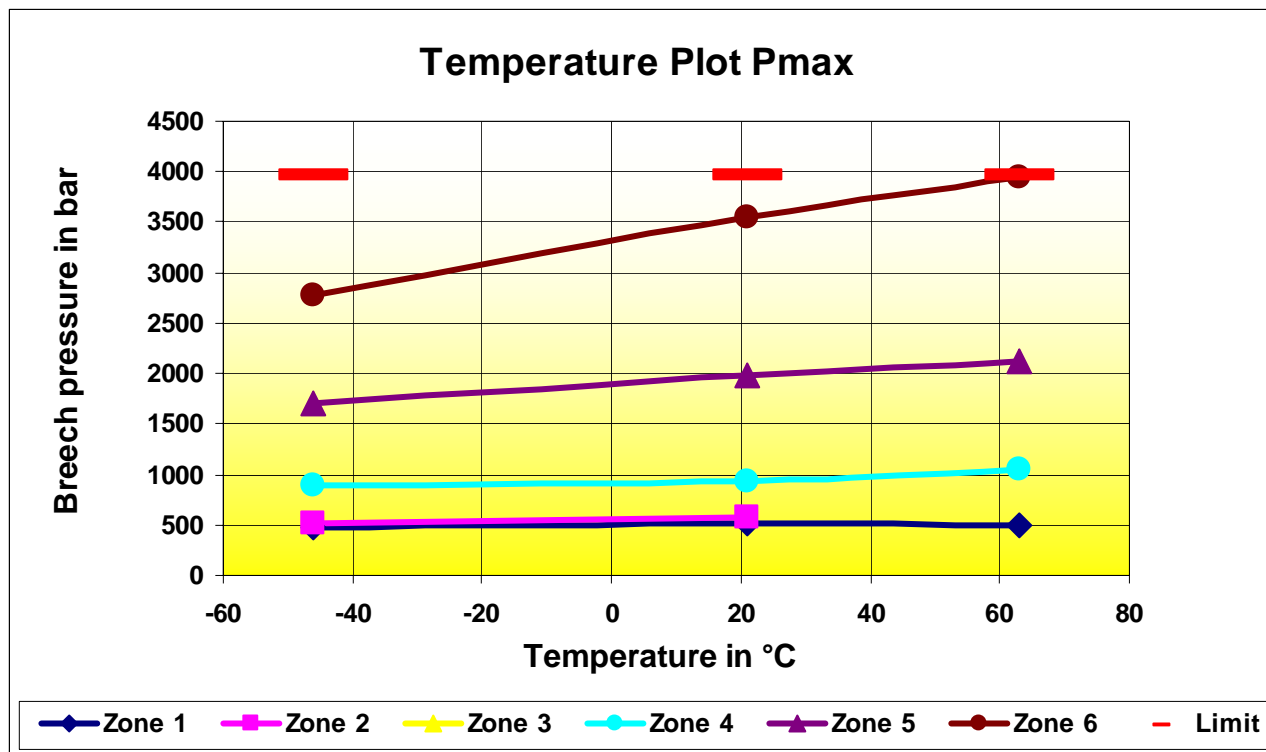
Zone 6 nominal (MIL) 633m/s

Achieved: 634.5m/s @ Zone 6 and +21°C

Not fired: Intermediate Zone 3

*Velocities close to targeted ranges*

## Results of Pressure Measurements (Piezo)



Pressure limit

57500psi

3965bar

*Pressure requirements achievable by correction of propellant design (known and reliable measures)*

### Pressure at hot close to permissible limit!

- Correction of temperature response for high zone!
- Modelling of propellant design (coating parameters)
- Optimization of pressure budget (headroom in charge weight)

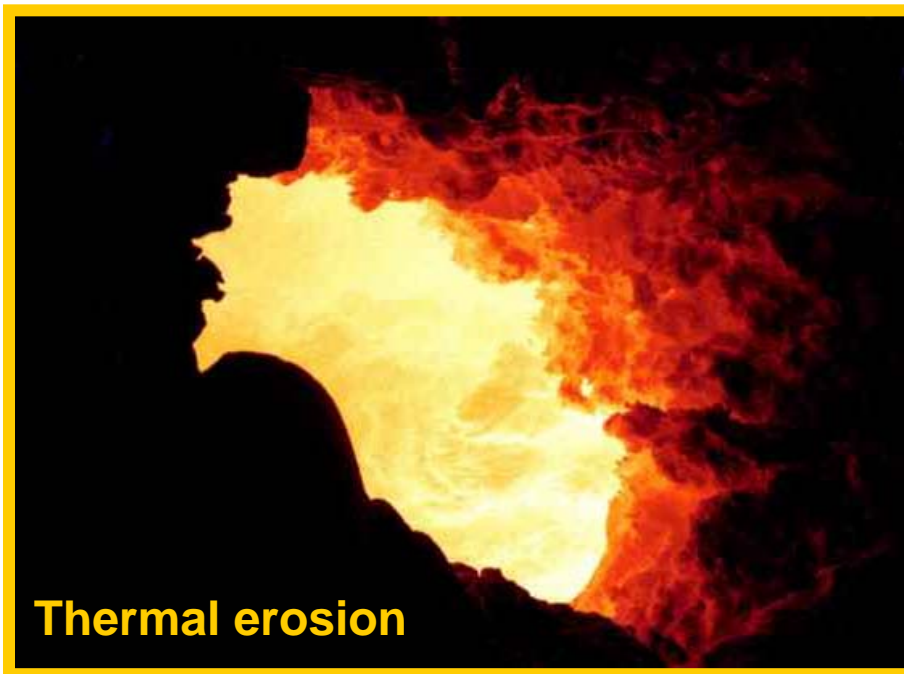
## Summary and Conclusions

- 6 Zone conception with propellant design based on same generic formulation (3 different grain types of ECL<sup>®</sup>)
- Required and targeted velocities for individual charge increments fulfilled
  - Range and overlap requirements
- Pressure budget for highest zone at hot (145°F) close to operational pressure limit
  - Temperature response of higher charge zones to optimize (adaptation of burn rate profile)
- Consistent pressure – time curves (no signs of pressure waves = safe for firing at any condition)
- Headroom for propellant charge for optimization of pressure budget

***Final Requirement achievable with slight modifications!***

## Two Major Influences on Barrel Wear

- Barrel wear due to Thermal Erosion
- Barrel wear due to Chemical Erosion



## Comparison ECL Charge Design vs fielded Design (results from YPG firing test June 2008)



### M67 Zone 7

M1 propellant (flame temperature **2575K**)

- Velocity 503.3m/s
- Pressure **39275psi**



### M200 (stand alone charge)

M30 propellant (flame temperature **3070K**)

- Velocity 651.5m/s
- Pressure **46081psi**



### ECL (Zones 5 and 6)

ECL propellant (flame temperature **2850K**)

- Velocity zone 5 509.6m/s
- Velocity zone 6 634.5m/s
- Pressure zone 5 **28698psi**
- Pressure zone 6 **50812psi**

## Calculation of Erosion (thermo-mechanical approach)

$$\text{Erosion} \sim (m_c)^{1.5} \cdot (T_{ex})^7 \cdot (v_0)^{1.4} \cdot (p_{max})^5$$

**Semi-empirical formula  
based on experience and  
measured values**

$m_c$	=	Charge Mass
$T_{ex}$	=	Flame Temperature
$v_0$	=	Muzzle Velocity
$p_{max}$	=	Peak Pressure

**Barrel erosion has been measured through life assessment and proof in tank and artillery guns.**

**Measurements in plain steel barrels with mechanical measurement and thin layer method (activated steel).**

## Barrel Life Estimation Assessments (Different comparisons)

$$\text{Erosion} \sim (m_c)^{1.5} \cdot (T_{ex})^7 \cdot (v_0)^{1.4} \cdot (p_{max})^5$$

Semi-empirical formula shows the main drivers for barrel erosion are **flame temperature** and **pressure level**

### M200 vs ECL zone 6

Theoretical; only flame temperature changes

- 41% less erosion

### M200 vs ECL zone 6

Practical; ECL not optimized; YPG results

- 7% less erosion

### M67 zone 7 vs ECL zone 5

Practical; ECL not optimized; YPG results

- 62% less erosion

*M200 still slightly more erosive compared to ECL<sup>®</sup> Zone 6!*

*M67 Zone 7 significantly more erosive compared to ECL<sup>®</sup> Zone 5!*

## Calculation of erosion (thermo-chemical approach: Lawton)

$$\text{Erosion} = A \exp(T_{\max} / B_o)$$

Theoretical approach with  
respecting gas composition

- **A:** Propellant erosion coefficient (depends on propellant gas composition)
- **B<sub>o</sub>:** Hardness coefficient (105 for typical gun steel)
- **T<sub>max</sub>:** Maximum bore temperature during firing (assumption 80% of flame temperature)
- $$A = \exp(0.23f(\text{CO}_2) + 0.27f(\text{CO}) + 0.28f(\text{H}_2\text{O}) + 0.74f(\text{H}_2) + 0.16f(\text{N}_2) + 1.55f(\text{R}) - 31.36)$$
- **f:** The volume fraction of each species in percent
- **f(R):** Represents the dissociated products

*Hydrogen as main cause for erosion (steel attack)*



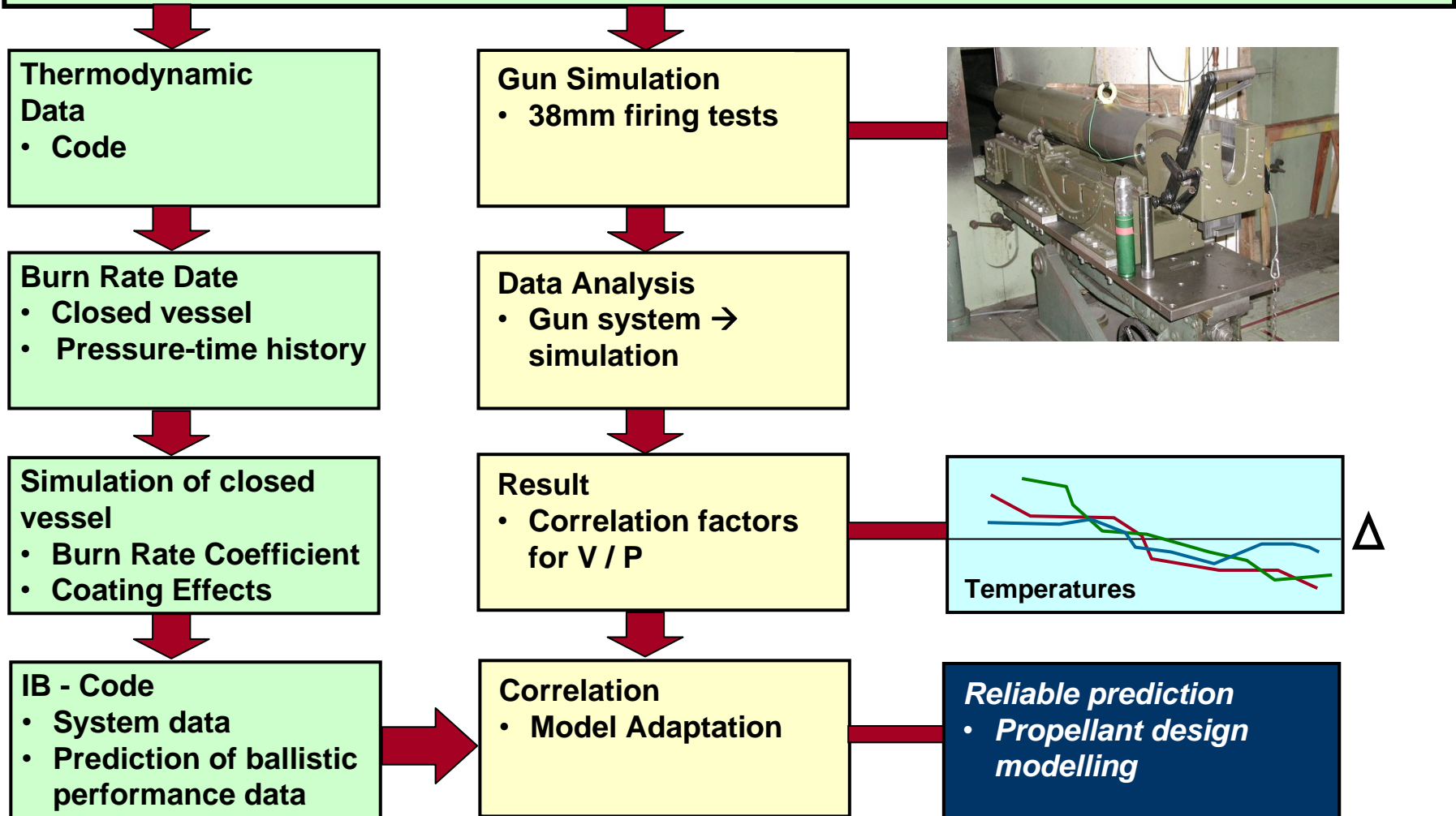
## Conclusions

- Thermal Erosion and Chemical Erosion have different aspects which have to be observed separately!
- Erosion calculated by thermo-mechanical approach shows good-natured behaviour of ECL propellant!
- Erosion compared to M67 charge (M1 propellant) expected to be significantly lower (pressure difference)
- Erosion compared to M30 (stand alone charge) at least comparable or slightly better for ECL propellant (lower flame temperature)!
- The thermo-chemical approach by Lawton results in comparable erosion due to comparable Hydrogen contents (for same pressure)!



## Modelling of Propellant Design

### Methodology for Simulation of Ballistic Data





**Adaptation of Propellant Design**

**Analysis of firing results (2009)**

**Reproduction in larger quantities**

**US Qualification ?**

- ARDEC for supporting ECL technology for this program
- Nguyen Tran from ARDEC for leading this project
- Peter Zoss and Kurt Ryf from Nitrochemie as co-workers from Nitrochemie
- Kelly Moran, Duncan Langlois and Steven Ritchie from ATK as our strategic partner

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