



SmallSat Conceptual Design Trade and Cost Modeling Tool

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***NDIA 12th Annual Systems
Engineering Conference***

***Systems Engineering
Development Environment***

October 29, 2009





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Advatech Company Overview

- **Founded in 1995, Owned/Operated by Aerospace Engineers**
- **Locations in California, Arizona and Virginia**
- **R&D and Engineering Services**
 - **Integrated tool development and analysis**
 - Space vehicle modeling
 - Launch vehicle design and cost
 - Hypersonic vehicles
 - Trajectory analysis
 - Range safety (Responsive Range Safety)
 - **Software design and development**
 - **Engineering design and analysis**
 - Structural
 - Thermal
 - Composites
 - Advanced space propulsion (electric / nuclear)
 - Tactical communications



Traditional Design Approach

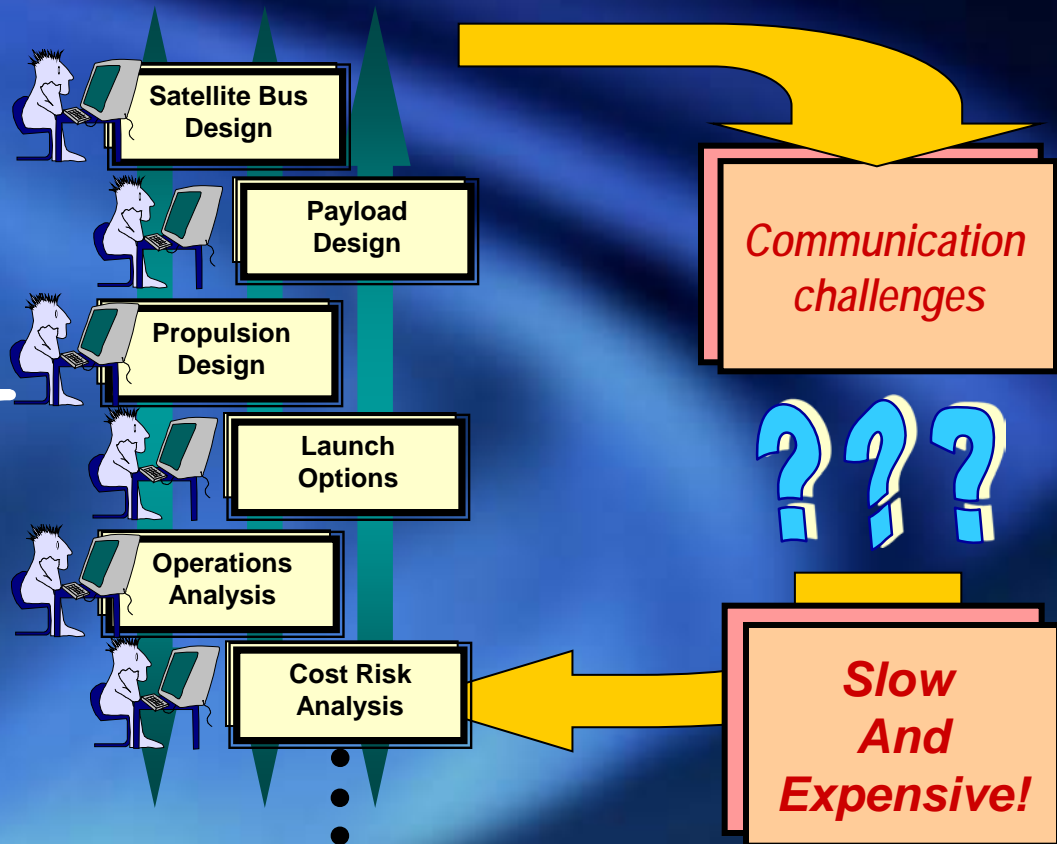
Early Design Challenges of High-Performance Complex Systems

What is the current design process?



System Requirements

- Mission requirements
- Size, weight, power
- New technology insertion
- Performance
- Schedule
-
-
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Integrated Design Approach

Solution to Early Design Challenges of High-Performance Complex Systems

Integrated optimizing tools provide:



System Requirements

- Mission requirements
- Size, weight, power
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Integration
Core Module

User
Interface
Module



Bus Module

Payload Module

Propulsion Module

Launch Module

Operations Module

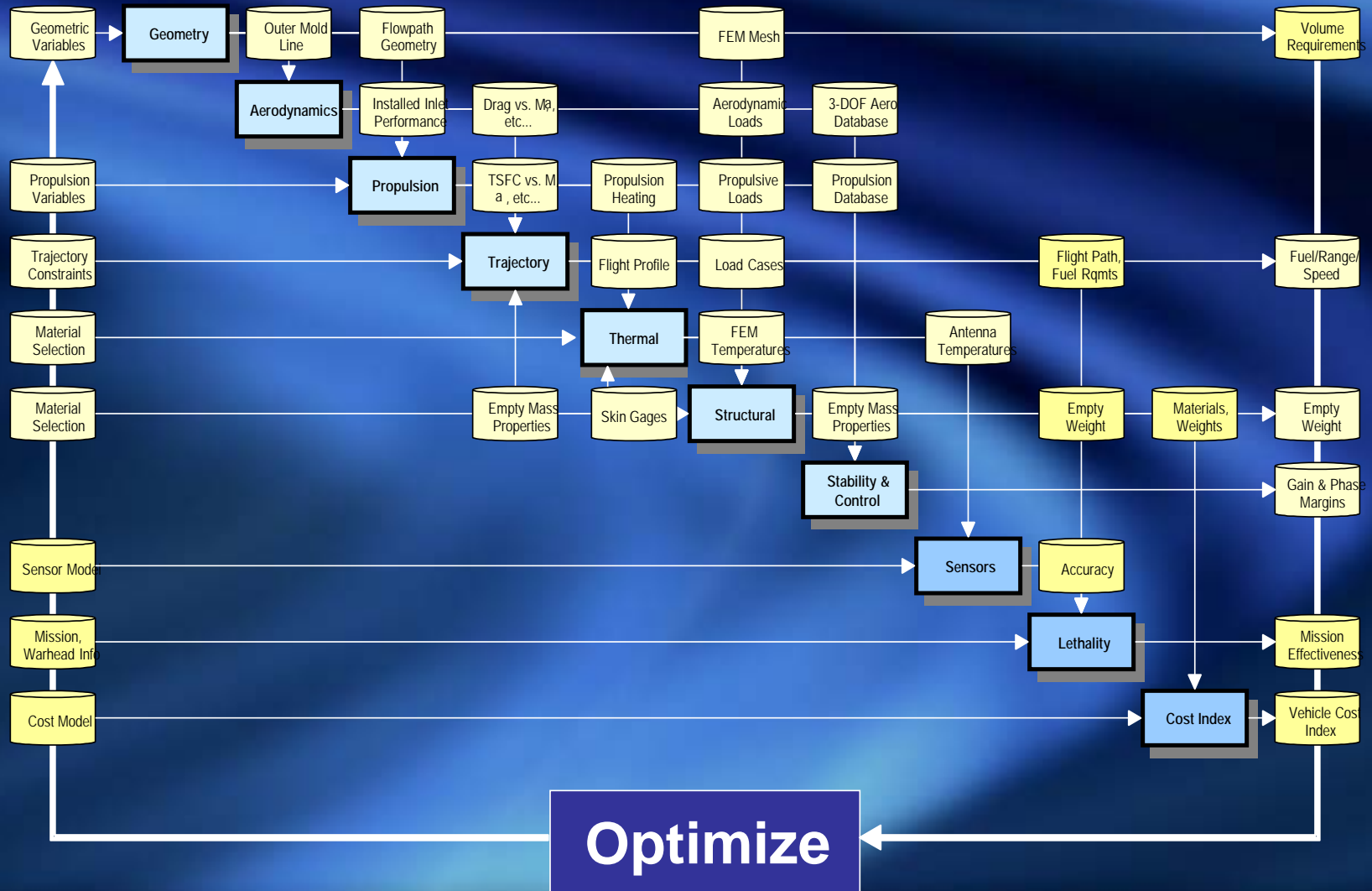
Cost/Risk Module

**In Days
Not
Weeks!**





Integrated Tool Suite Process Flow





Integrated Tool Suite Benefits

- Continuous trade study capability throughout the acquisition life cycle
- Iteratively model key parameters in the pre-system acquisition phase
 - ConOps
 - Performance
 - Cost
 - Schedule
 - Technology risk
- Responsive turnaround – days, not weeks!
- System and subsystem trade analysis
- Continuous knowledge capture and update

Fundamental enabler for building the best performing system within the cost, schedule and technology constraints



Advatech Integrated Projects

- Space vehicle design and cost (ACES-ISET)
- Advanced Cost Model (ACM)
- Launch vehicle design, operations and cost (IPAT)
- Hypersonic aeromechanics tool (IHAT, FPAT)
- Integrated Physics Based Cost / Risk Analysis Tool (ICAT)
- Composite Rotor Blade and Wing Structural Design Tool
- Component Integrated Modeling Simulation and Test Analysis Environment (CIMSTA)
- Naval Engineering Analysis Tool (NEAT)
- Virtual Satellite Integration Effort (VSIE)
- Small Satellite Launch Vehicle (SPRITE)
- Analytical Methods for Sandwich Core Termination
- Integrated High Payoff Rocket Propulsion Technology (IHPRPT)
- Aircraft Vulnerability Model (AVM)
- Combined Hall Effect Thruster Code (CHETC)
- Field Reverse Configuration (FRC) Thruster Model – Orbit Transfer Vehicle System Model
- Highly Mobile Tactical Communications (HMTC)
- Integrated Solid Motor Analysis Tool (ISMAT)

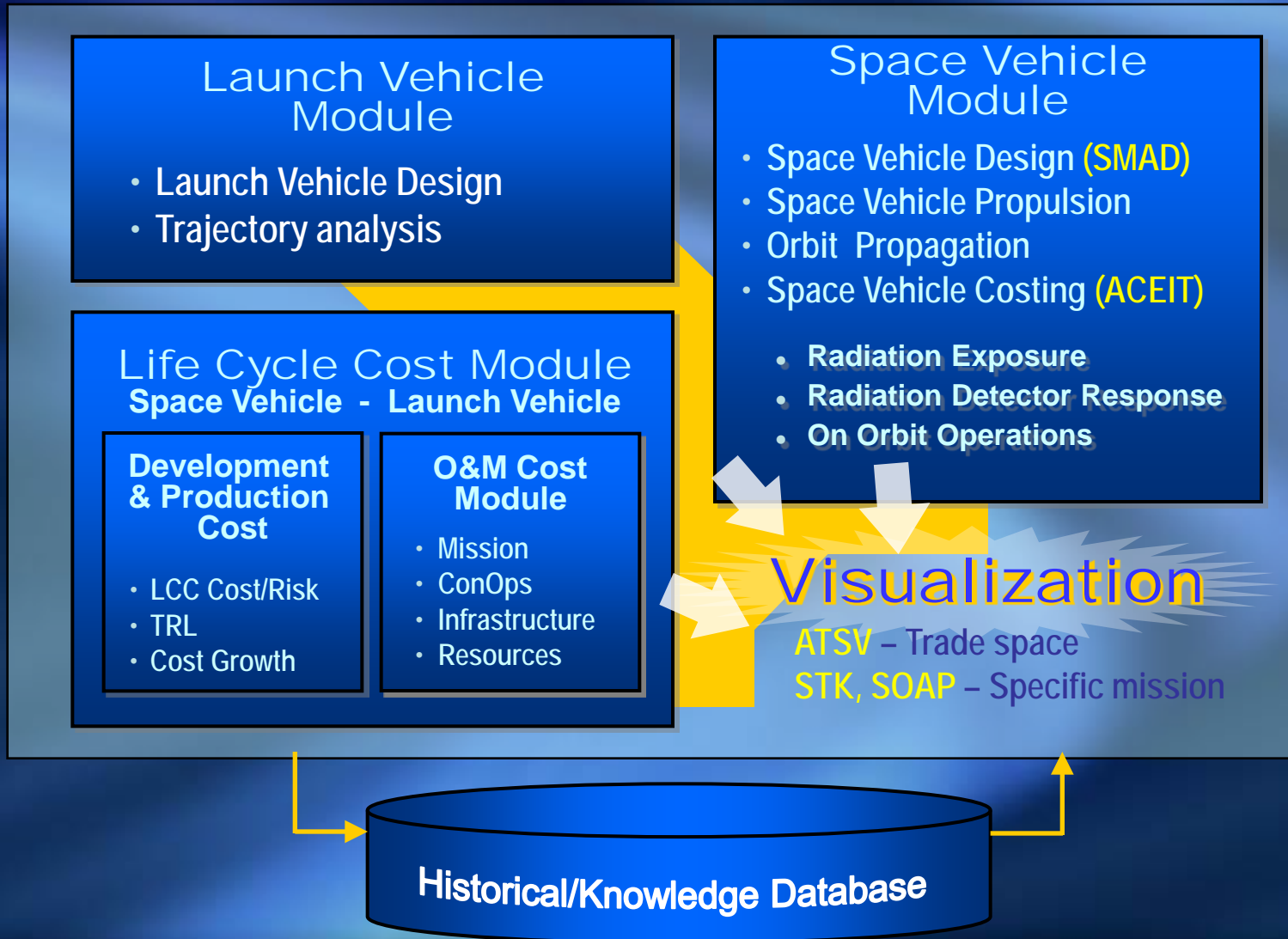


SmallSat Conceptual Design Tool

- **Advanced Computational Engineering Simulator – Integrated Space Analysis Tool (ACES-ISET)**
- Customer: Air Force Research Laboratory, Space Vehicles Directorate, Kirtland AFB, NM
- Partners: Tecolote Research, MCR LLC, RSSI
- An integrated, multi-disciplinary engineering tool suite
 - Optimizes the design and cost of space vehicles
 - Models the space environment
 - Selection of launch vehicles and modeling launch operations
 - Perform mission planning trade studies
 - Visualization of results

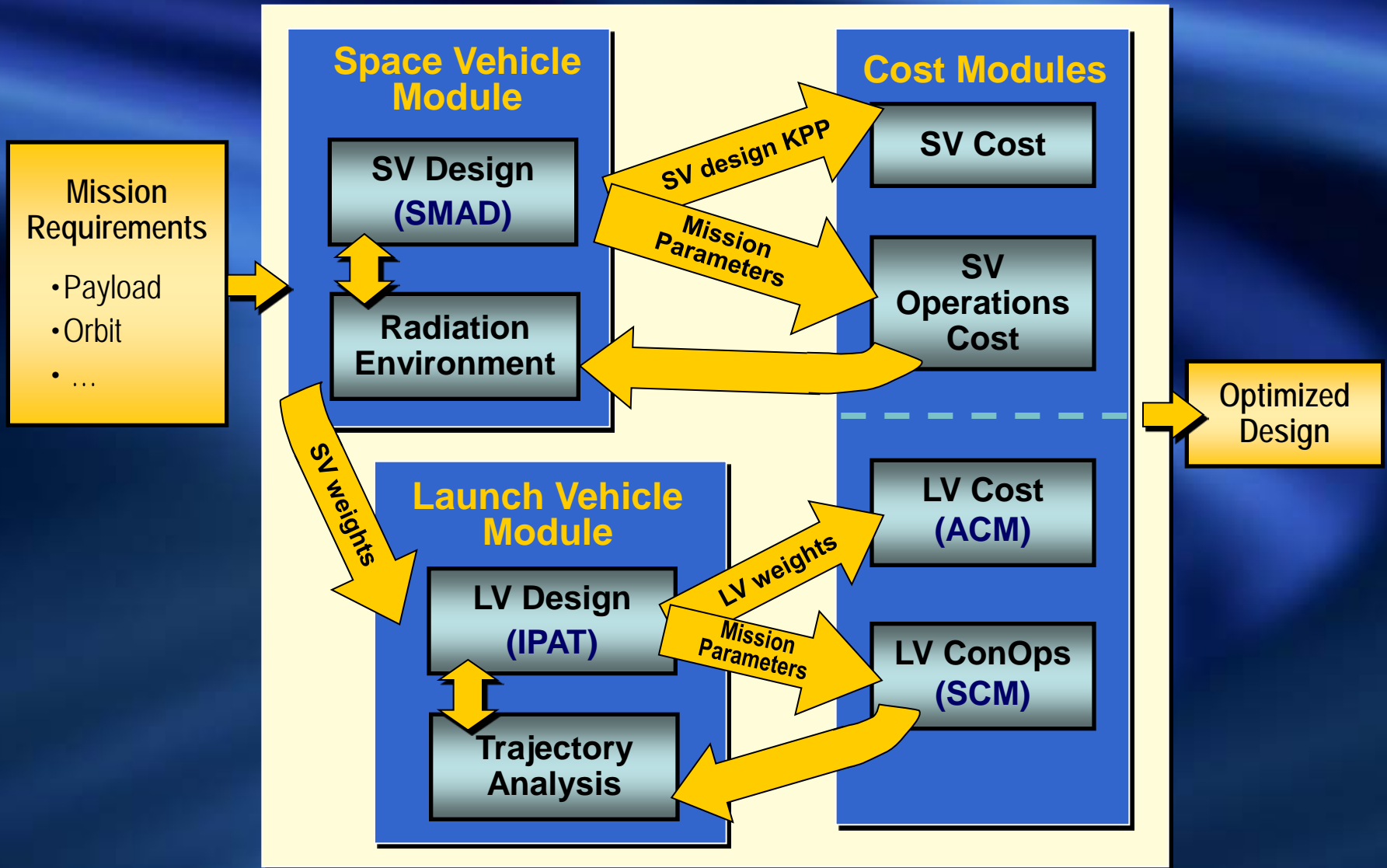


Integrated System and Cost Model (ISCM) - Tool Suite





ACES Conceptual Design





Cost Estimating

- **Cost modules fully integrated with design tools**
- **Cost estimating relationships based on**
 - historical data
 - sub-system weights
 - materials
- **Historical data used to identify cost growth rates related to Technical Readiness Levels (TRL)**
- **Cost and schedule are related to TRL and system engineering milestones**
- **Built in risk estimating capabilities**



Cost Risk Estimates

Cost growth incurred as technology matures



Risk

Triangular bounds (L,M,H) on weights drive S-Curves

Determined using FRISK, a deterministic risk analysis tool

S-Curves shift to the right with cost growth



Applications

- **Examples of Trade Studies**

- Effect of subsystem reduction on total vehicle design
- Concept evaluations of proposed TacSat-5 concepts
- Cost impact of alternative TacSat-3 designs
- Launch vehicle selection and cost for satellite constellation
- Trajectory analysis for DSX alternative orbits
- Concept modeling for ORS modular satellite architectures



Sensitivity Study

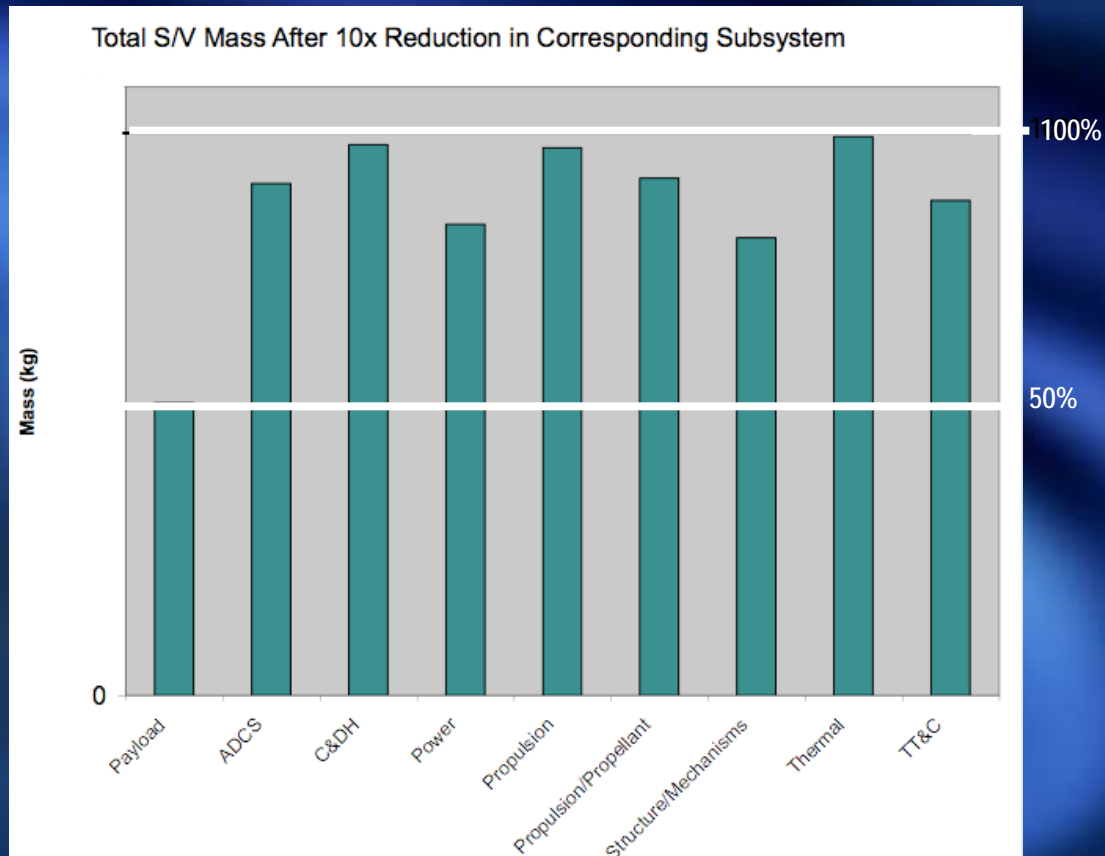
- Study highlights

- Quantitative and qualitative data on the impact of decreasing Size, Weight and Power (SWAP) of individual subsystems on the overall space vehicle SWAP
- Insight on space vehicle subsystems & components interaction
- Determined feasibility of reducing Space Vehicle mass by factor of 4
 - **Only through cross-subsystem functionality**
- Identified two major areas for future focused research
 - **energy conversion**
 - **structural materials**
- Presented at the 6th Responsive Space Conference 2008

Expectations for return on research investments can be bound by quantifying system-level effects of a single breakthrough



Sensitivity Study – Sample Results



Effect of subsystem 10x mass reduction on total space vehicle mass
Each bar represents the effect of a single subsystem mass reduction



TacSat-5 Concept Evaluations

- Study highlights
 - Source solicitation evaluation
 - Tiger Team approach

Integrated Design Approach

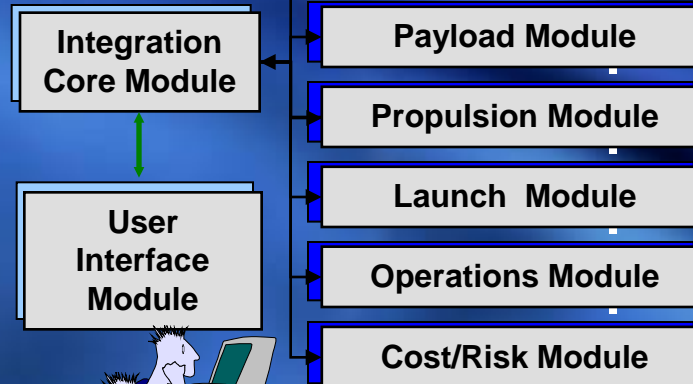
Solution to Early Design Challenges of High-Performance Complex Systems

Integrated optimizing tools provide:



System Requirements

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***In Days
Not
Weeks!***





TacSat-5 Concept Evaluations

● Study highlights

- Source solicitation evaluation
- Tiger Team approach
- Thirteen concepts evaluated (classified and unclassified)
- Identified and quantified issues with concept proposals:
 - Projected costs that exceed available budget
 - Costs that assume payload design heritages not supported in proposal
 - High risk payloads based on low TRLs, long development schedules or payloads exceeding mass budget limits
- Identified and quantified issues with transitioning to operational version
- Demonstrated that some proposals/CONOPS
 - Contained inconsistent assumptions
 - Contained questionable assumptions that needed further investigation

Knowledge gained during conceptual design phase enabled decisions about designs and mission capabilities before a large investment was committed



Alternative TacSat-3 Designs

- Study highlights
 - Ongoing study
 - Determine subsystem design changes needed to create an “operationalized” version of TacSat-3.
 - Model and evaluate design modifications
 - Baseline design
 - Payload reductions
 - Increased mission length
 - Subsystem redesign with newer technologies
 - Determine cost of design modifications
 - Select and determine procurement costs of launch vehicles needed to launch a satellite constellation

Cost estimates for design modifications are affected by subsystem heritage and technology maturity.



Lessons Learned

- **Selecting integration environment**
 - License cost
 - Performance (speed)
 - Portability (platforms)
 - Flexibility and ease of development
 - Scalability
 - Automated parameter management (facilitates trade studies)
 - User interface
- **Selecting M&S tools to be integrated**
 - Existing customer tools
 - Validation level (industry accepted)
 - OTS versus development
- **Data availability and reliability**
 - Proprietary data
 - Validation level
- **Export control and use restrictions**
- **Managing customer expectations**



Conclusions

- **Integrated tools suites**
 - **provide substantiated, traceable and reproducible results**
 - **reveal interdependencies of cost, risk, schedule, and performance**
 - **provide higher confidence in cost and schedule estimates**
 - **enable better management of technology investment by decision makers**

Concepts and processes are applicable to design domains beyond space