



U.S. Army Research, Development and Engineering Command



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The Malcolm Baldrige National Quality Award logo is a large, stylized gold ribbon with a red star in the center, set against a dark red background with a faint grid pattern.

***TECHNOLOGY DRIVEN. WARFIGHTER FOCUSED.***

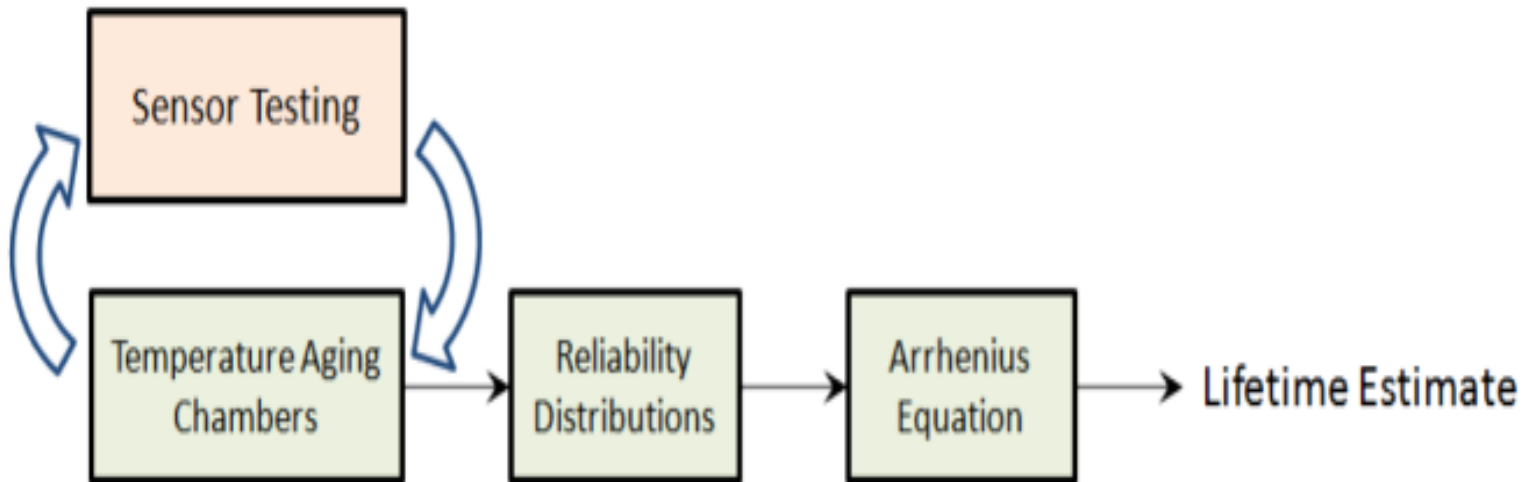
***Determination of the Shelf Life of MEMS Navigation-Grade Sensors through Use of Accelerated Aging Principles***

***April 14, 2011***

***James A. Sarruda***



- Determine the life expectancy of a generic micro electro-mechanical systems (MEMS) inertial measurement unit (IMU) designed specifically for the use in a precision munition.



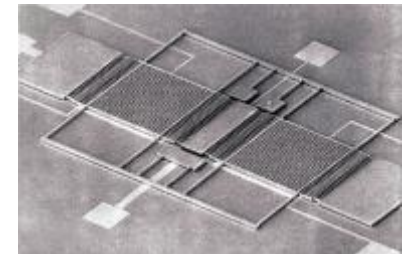
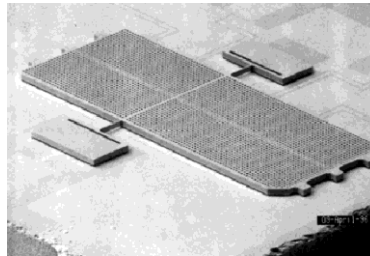


- Degradation in performance parameters of a MEMS IMU over time is not well understood.
  - Natural aging not an option
  - Developing navigation capability
- Artificial aging is a viable option to understand product reliability.
  - Precision munitions have been expensive
  - Guidance and Control is a key cost driver





- Guided munition weapon systems are in need of extremely miniaturized, navigation-grade sensors in order to increase the lethality of the Army's precision guided products.
  - Complex environments pose a severe challenge to the warfighter, and therefore the advancement of key technology is necessary to meet stringent user requirements.
  - MEMS IMU provides full six-degrees-of-freedom triaxial motion measurements.
  - All indirect fire systems substantially benefit from precision capability.

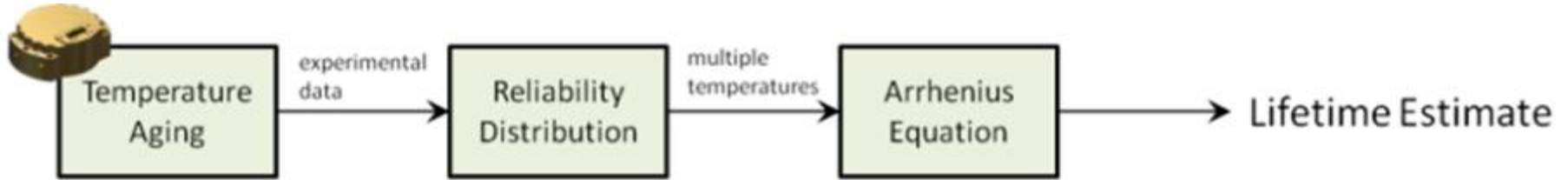




- Testing is designed to obtain the life expectancy of the inertial sensors within the inertial measurement unit system, not the life expectancy of the IMU system as a whole.
  - Performance independent of sensor integration
  - Greater sample size
- Reliability Prediction
  - A “time to failure” population distribution is required to predict shelf life
  - Weibull distribution is commonly used in reliability engineering
- Success Criteria
  - Sensors must “fail” for lifetime equations to be appropriate.
  - Sensor performance outside program goals will be considered a failure.







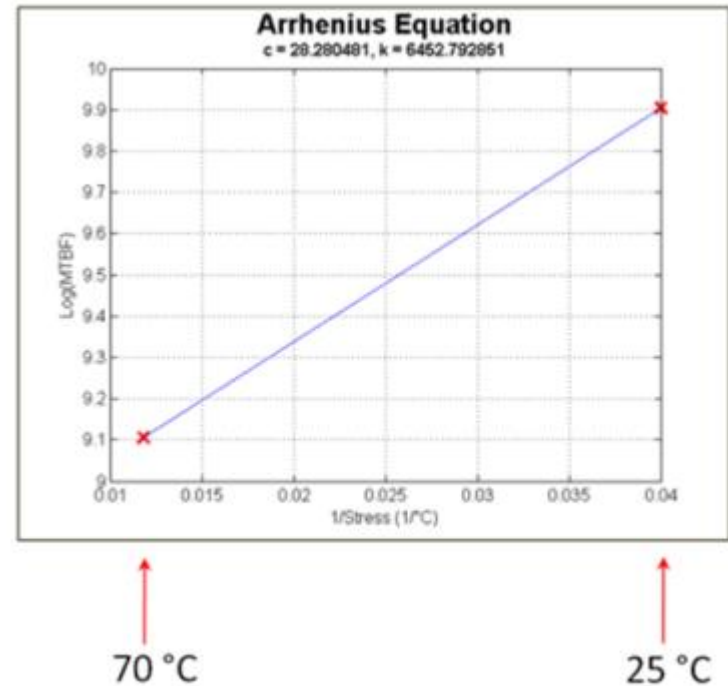
- Arrhenius equation correlates lifetime to temperature stress.

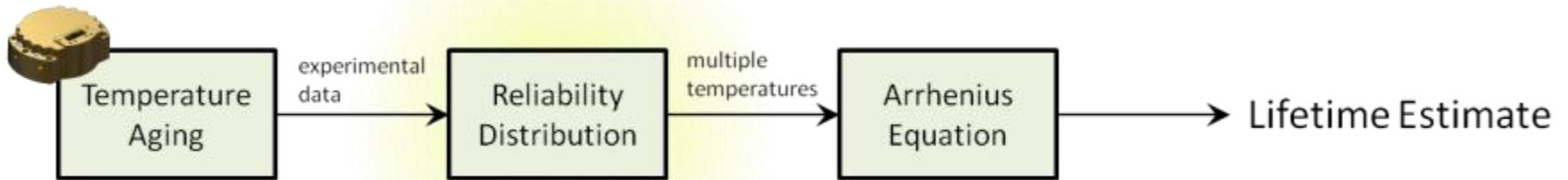
$$t = ke^{\frac{c}{S}}$$

Where

t is the time to failure for a percentage of the population  
 k and c are constants  
 S is the amount of stress

- At a minimum, two temperature age estimates are needed to estimate the equation's parameters.

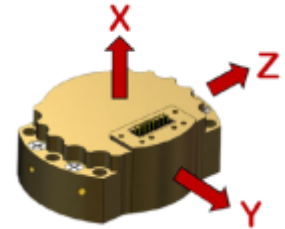




- A “time to failure” population distribution (reliability distribution) is required to use the Arrhenius equation.
- A reliability distribution is selected from a best fit to the experimental data failure times.
- The sensor parameters must fail for the reliability distribution to be determined.
- Weibull distribution will be utilized for this testing as this distribution is commonly used in reliability engineering.



- Each inertial IMU contains an inertial sensor assembly (ISA) with six inertial sensors.
  - Angular rates are measured using three orthogonal MEMS gyroscopes along the x, y, and z axes.
  - Linear acceleration is measured using three orthogonal MEMS accelerometers along the x, y, and z axes.



Name	Symbol <sup>1</sup>	Units	Name	Symbol <sup>1</sup>	Units
Angle Random Walk (ARW)	$g_{i,ARW}$	$^{\circ}/\sqrt{hr}$	Velocity Random Walk (VRW)	$a_{i,VRW}$	$m/s/\sqrt{hr}$
Bias Error	$g_{i,Berr}$	$^{\circ}/hr$	Bias Error	$a_{i,Berr}$	mg
Bias Instability	$g_{i,Bstab}$	$^{\circ}/hr$	Bias Instability	$a_{i,Bstab}$	mg
Turn-on to Turn-on Bias Error Repeatability	$g_{i,Brep}$	$^{\circ}/hr$	Turn-on to Turn-on Bias Error Repeatability	$a_{i,Brep}$	mg
Self-Heating Settling Time	$g_{i,setT}$	hr	Self-Heating Settling Time	$a_{i,setT}$	hr
Scale Factor Error	$g_{i,SFerr}$	PPM	Scale Factor Error	$a_{i,SFerr}$	PPM
Scale Factor Asymmetry	$g_{i,SFSym}$	PPM	In-Run Scale Factor Stability	$a_{i,SFir}$	PPM
Scale Factor Nonlinearity	$g_{i,SFlin}$	PPM	In-Run Bias Stability	$a_{i,Bir}$	mg
In-Run Scale Factor Stability	$g_{i,SFir}$	PPM	Sensor Misalignment	$a_{\delta i}$	$\mu rad$
In-Run Bias Stability	$g_{i,Bir}$	$^{\circ}/hr$			
Sensor Misalignment	$g_{\delta i}$	$\mu rad$			







- Shape and scale parameters will be determined via linear regressions.

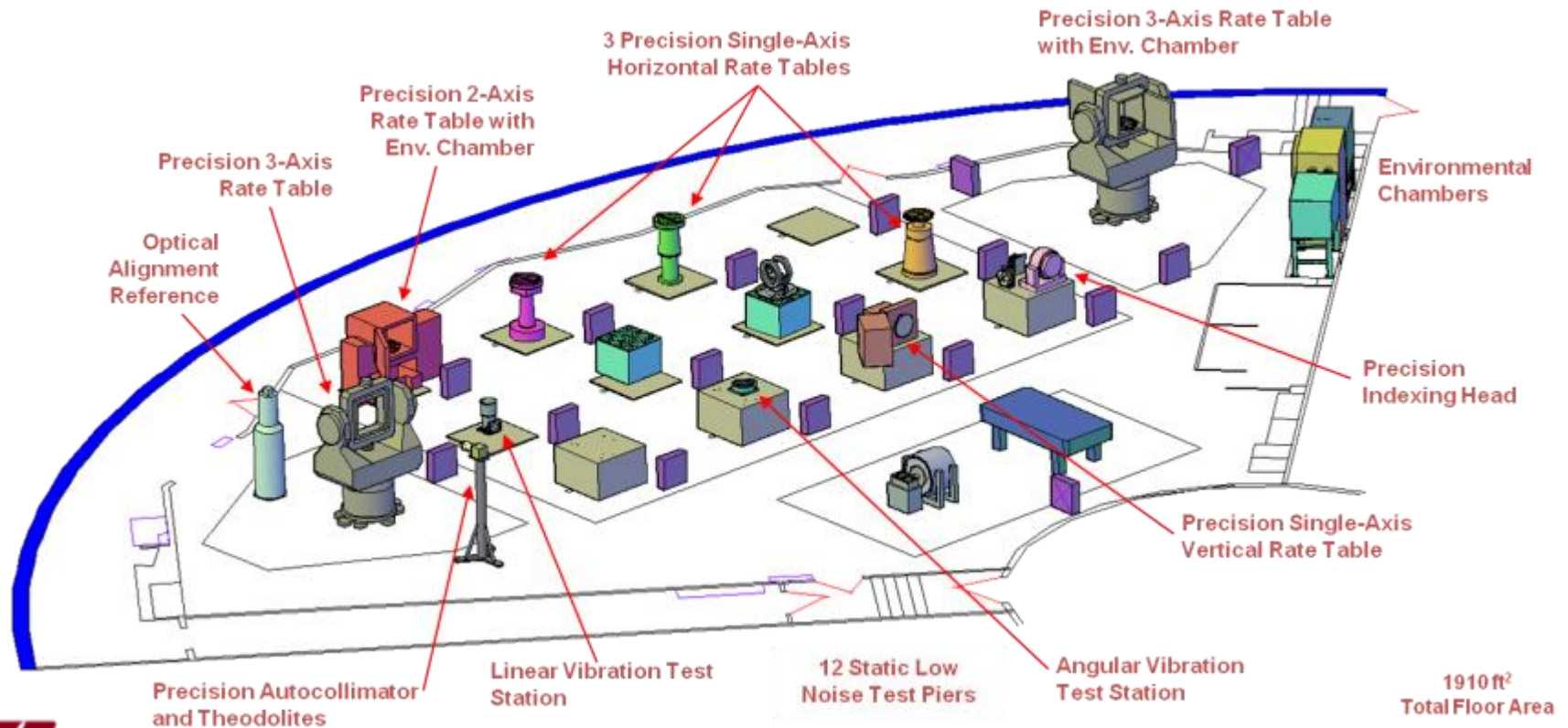
Gyros Parameter	Weibull Reliability Distribution Parameters	25 °C	70 °C	Arrhenius Constants
Bias Error	Shape ( $\beta$ )			K = , C =
	Scale ( $\theta$ )			
Scale Factor Error	Shape ( $\beta$ )			K = , C =
	Scale ( $\theta$ )			
Non-Orthogonality	Shape ( $\beta$ )			K = , C =
	Scale ( $\theta$ )			

Accelerometer Parameter	Weibull Reliability Distribution Parameters	25 °C	70 °C	Arrhenius Constants
Bias Error	Shape ( $\beta$ )			K = , C =
	Scale ( $\theta$ )			
Scale Factor Error	Shape ( $\beta$ )			K = , C =
	Scale ( $\theta$ )			
Non-Orthogonality	Shape ( $\beta$ )			K = , C =
	Scale ( $\theta$ )			





- Provides capabilities to conduct research as well as test the most precise navigation sensors and equipment
- Building design reduces noise internal to the building itself



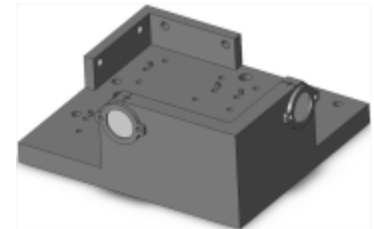
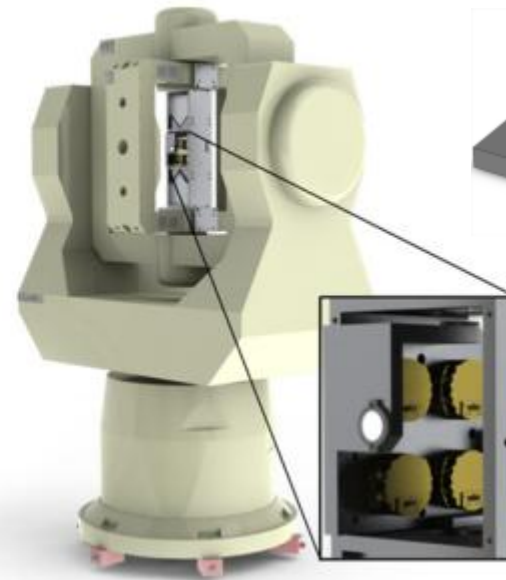
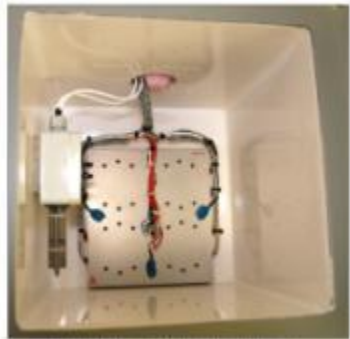


- **Artificial Aging**

- The temperature chamber provides air temperature, pressure, and humidity data.
- Mounting fixture stabilized at target temperature during aging.

- **3-Axis Characterization**

- Two alignment mirrors allow sensors to be precisely oriented on the fixture to the North, East, and to gravity.
- No need to remount sensors.





- Static Test
  - Sensors are powered on and held motionless for a period of 30 minutes.
- Rate Test
  - The sensor input axis of the gyros under test is positioned along the table's axis of rotation.
  - Sensor data is collected ten times according to a rate profile.
- Tumble Test
  - The sensor input axis of the accelerometers under test were positioned along north.
  - Sensor data is collected five times according to a tumble profile.

1	Static – X axis
2	Static – Y axis
3	Static – Z axis
4	Rate – X axis
5	Rate – Y axis
6	Rate – Z axis
7	Static – X axis
8	Static – Y axis
9	Static – Z axis
10	Tumble – X axis
11	Tumble – Y axis
12	Tumble – Z axis
13	Static – X axis
14	Static – Y axis
15	Static – Z axis





# Rate Profile



Table 16: Single Rate Profile

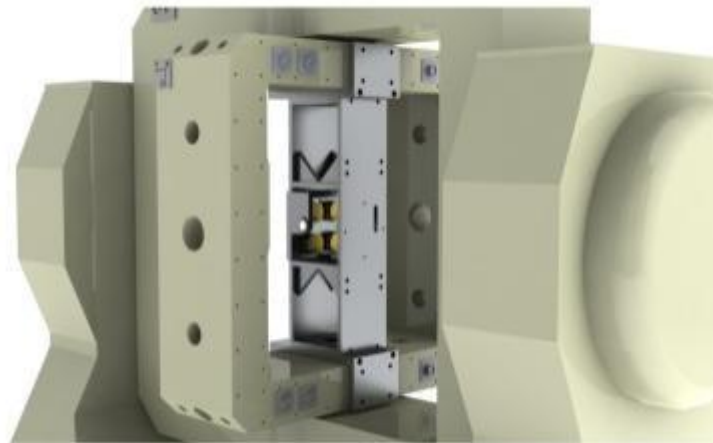
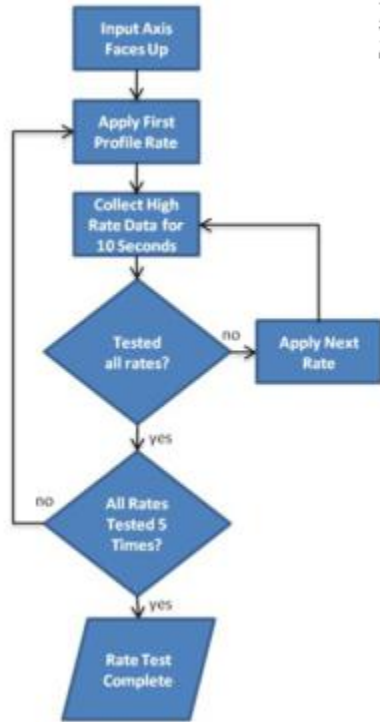
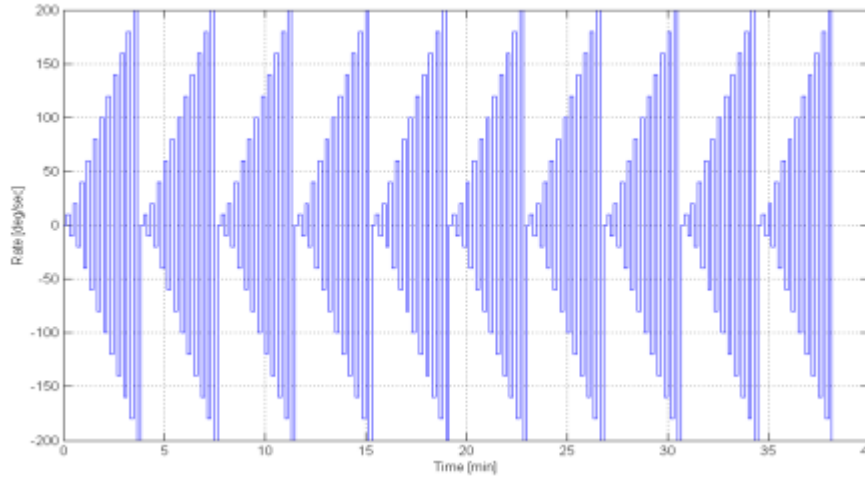


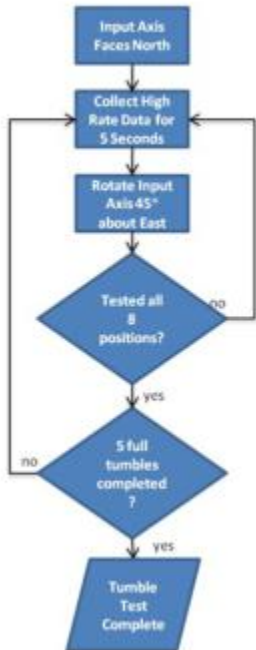
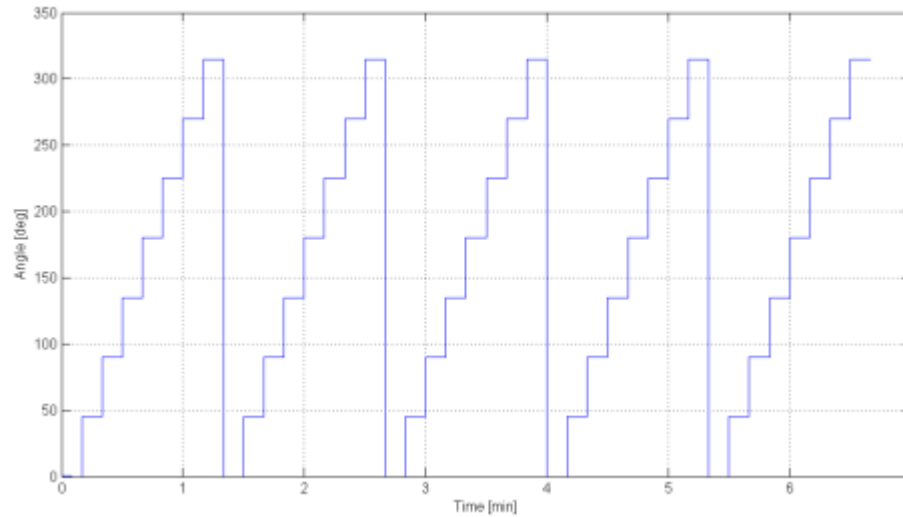
Table Rate deg/sec	Dwell Time sec
0	10
±10	10
±20	10
±40	10
±60	10
±80	10
±100	10
±120	10
±140	10
±160	10
±180	10
±200	10







# Tumble Profile



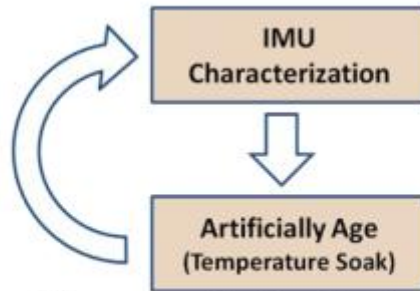
	Angle deg	Input Axis Orientation
1	0°	North
2	45°	
3	90°	Up (+g)
4	135°	
5	180°	South
6	225°	
7	270°	Down (-g)
8	315°	







- Initial characterization to baseline sensor performance parameters.
- Monthly characterization at the beginning of each month.
- Final characterization at the end of the 14 month period.



*Illustration 45: Test Cycle*

Year	Month	Action
2010	May	Test Setup
	June	Baseline Characterization
		Age
	July	Characterization
		Age
	August	Characterization
		Age
	September	Characterization
		Age
	October	Characterization
		Age
	November	Characterization
Age		
December	Characterization	
	Age	
2011	January	Characterization
		Age
	February	Characterization
		Age
	March	Characterization
		Age
	April	Characterization
		Age
May	Characterization	
	Age	
June	Characterization	
	Age	
July	Characterization	
	Age	
August	Final Characterization	





- Ongoing effort to be completed in September 2011
- Complete results are not available at this time
- Future MEMS IMU lifetime predictions can leverage off the process demonstrated in this testing program





- Increase the determination of reliability distributions by testing more IMUs.
  - Eight IMUs at each temperature step
- Additional temperature steps
- Further increase the aging process by increasing the aging temperature
  - More accurate lifetime prediction
- **Point of Contact**

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Picatinny Arsenal, NJ



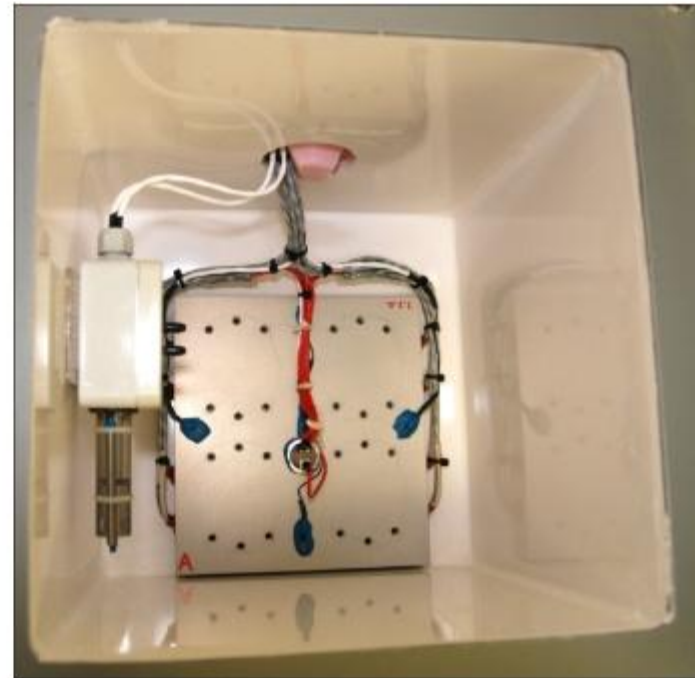
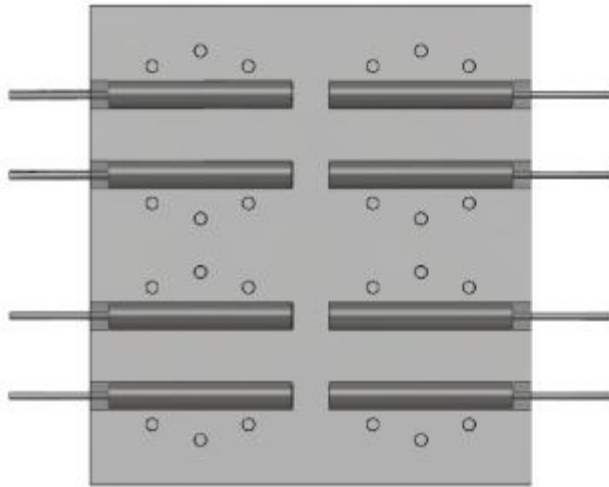


- Two thermal chambers and support electronics.
- IMUs are mounted to a custom fixture in each chamber .



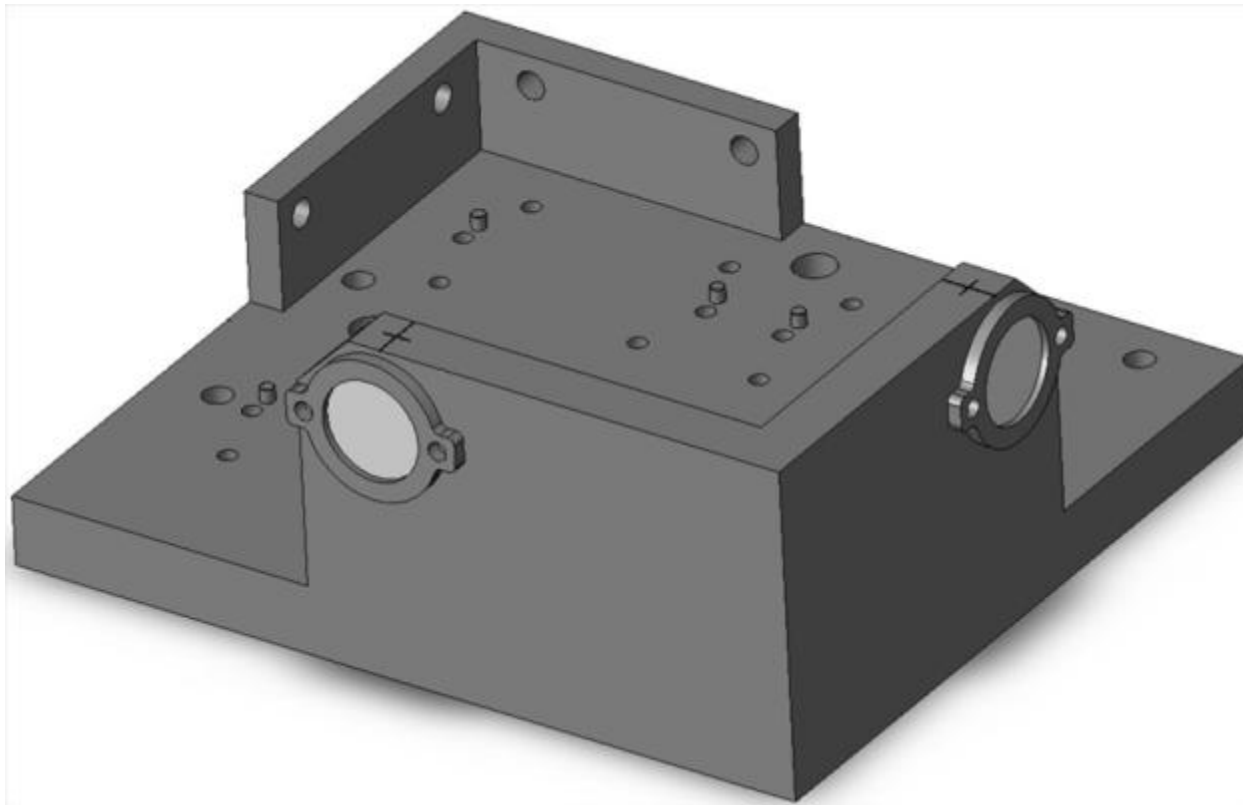


- The temperature chamber provides air temperature, pressure, and humidity control back to the support computer.

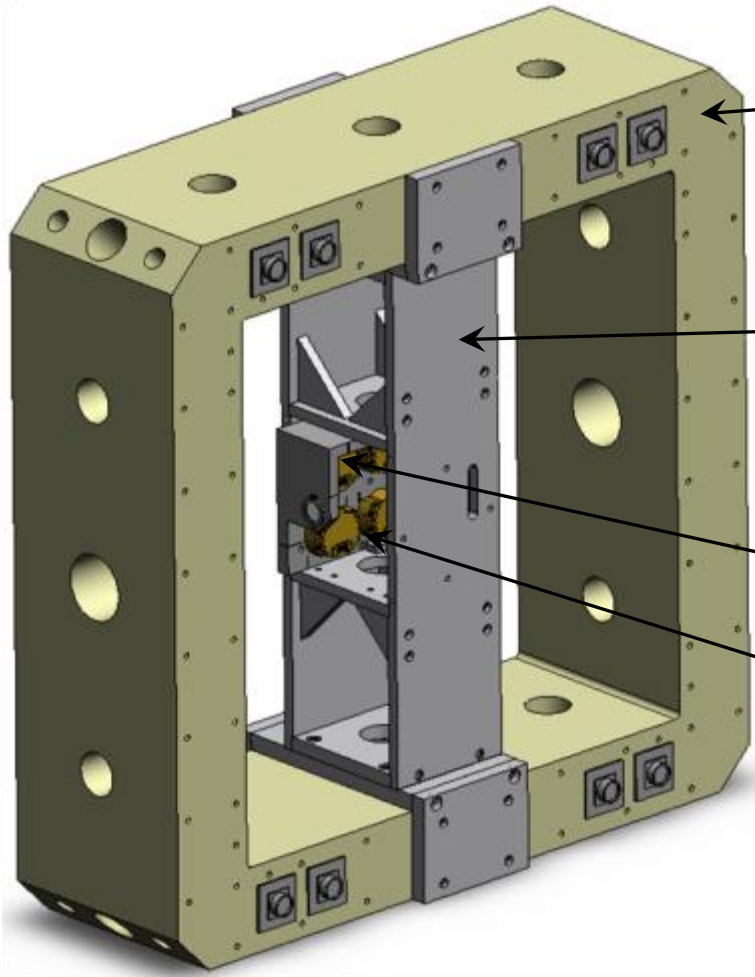




- Two alignment mirrors allow sensors to be precisely oriented on the fixture to the North, East, and Down.





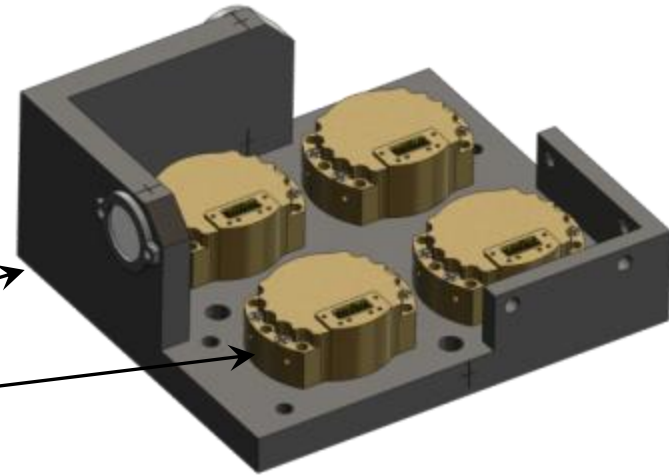


3-Axis Table  
Inner Gimbal

3-Axis Table  
Mounting Fixture

IMU Fixture

IMUs





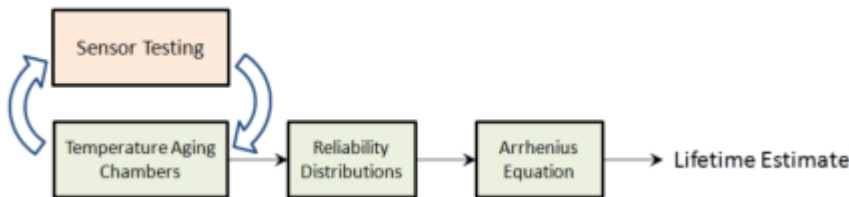
# Backup - Accelerated Aging Principles



- The Arrhenius equation can be generalized to correlate any stress to a population failure.
  - At minimum, two temperature age estimates are needed to estimate the equation's parameters.
- The devices under test do not fail simultaneously.
  - Weibull distribution used to model reliability

$$K = K_0 e^{-\frac{E}{BT}}$$

$K_0$  is the reaction rate at a known temperature  
 E is the activation energy of the reaction  
 B is Boltzmann's constant



$$R(x) = e^{-\left(\frac{x}{\theta}\right)^\beta}$$

R(x) is the population reliability at time x  
 θ is the Scale parameter  
 β is the Shape parameter





- Direct measurements of aging degradation can take decades.
- The Arrhenius model describes the effect temperature has on chemical reactions.

$$K = K_0 e^{-\frac{E}{BT}}$$

$K_0$  is the reaction rate at a known temperature

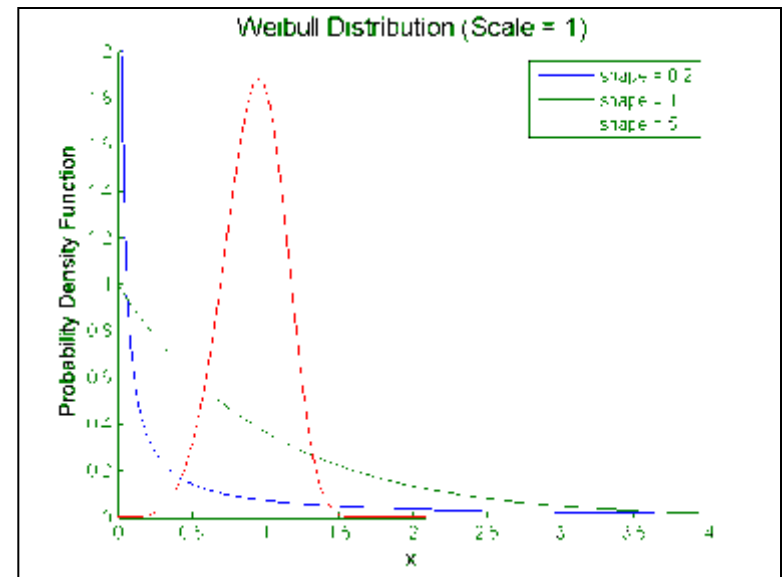
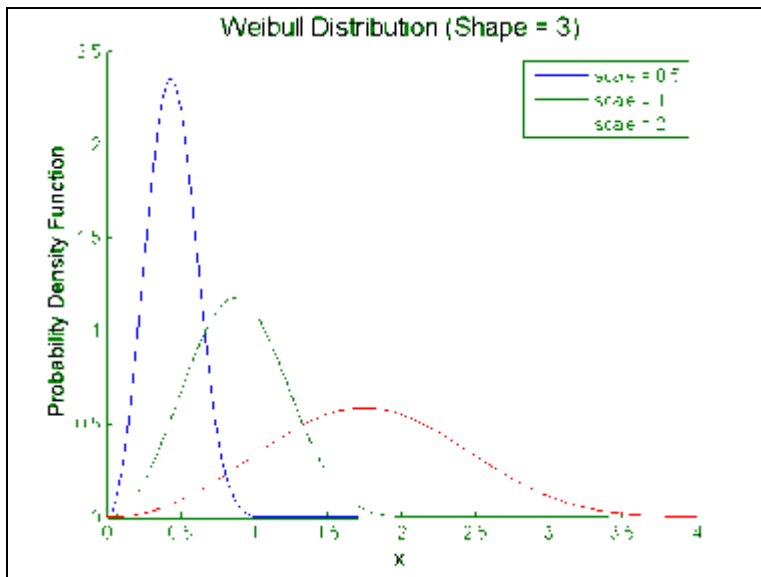
E is the activation energy of the reaction

B is Boltzmann's constant

- It is possible to induce artificial aging in a material or device by applying some form of stress.
- By calculating the reliability distributions at differing stress levels, the Arrhenius model can be utilized to calculate the failure rate at intermediate stresses.



Weibull distribution has two parameters to be calculated from the “failed” parameters data: Shape ( $\beta$ ) and Scale ( $\Theta$ )



$$f(x) = \frac{\beta}{\theta} \left(\frac{x}{\theta}\right)^{\beta-1} e^{-\left(\frac{x}{\theta}\right)^\beta}$$





- An appropriate out-of-spec value was chosen for each parameter
- The values are based upon the Common Guidance Common Sense (CGCS) program goals
  - Some of the sensor values were out of CGCS program thresholds at the start of the program
- Some of the gyro parameters are shown here

Gyro Parameter	Maximum Threshold
Bias Error	$\pm 100$ °/hour
Scale Factor Error	$\pm 400$ PPM
Non-orthogonality	$\pm 1400$ $\mu$ rad



- From the extrapolated parameter failure times, the shape and scale parameter for the Weibull distributions were determined via linear regressions on the data

Gyros Parameter	Weibull Reliability Distribution Parameters	25 °C	70 °C	Arrhenius Constants
Bias Error	Shape ( $\beta$ )	1.04	0.97	K = 0.86, C = 70.00
	Scale ( $\theta$ )	14.07	2.33	
Scale Factor Error	Shape ( $\beta$ )	1.18	1.34	K = 0.20, C = 94.32
	Scale ( $\theta$ )	8.81	0.78	
Non-Orthogonality	Shape ( $\beta$ )	0.61	1.27	K = 5.28, C = 1.12
	Scale ( $\theta$ )	459.76	26.04	

- From the calculated Weibull distributions, the population reliability for each parameter across temperature can be computed

