

Air Force Materiel Command

Developing, Fielding, and Sustaining America's Aerospace Force

High-Fidelity Physics- Based Modeling in Support of Test & Evaluation



U.S. AIR FORCE

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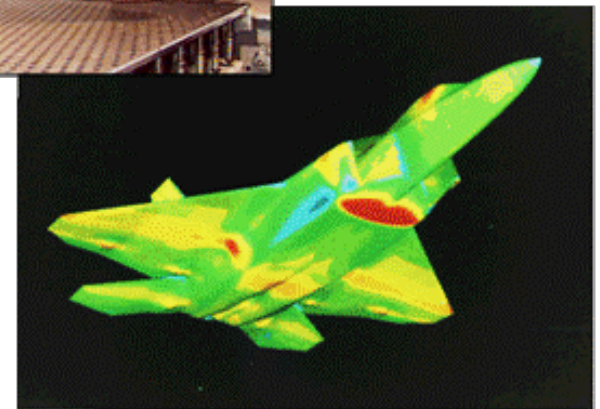
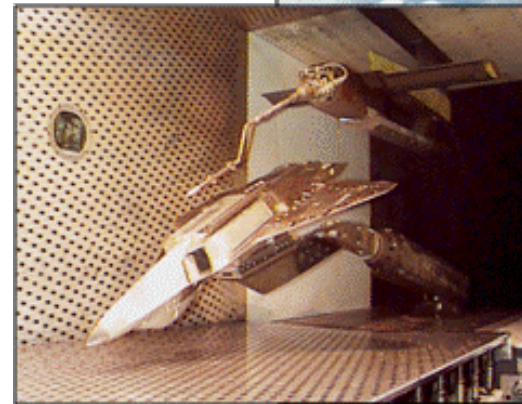
Integrity - Service - Excellence



Objectives



- Illustrate the history, diversity and intensity of physics-based modeling in T&E
- Indicate a future vision for increased utility of physics based modeling integrated with T&E in support of weapon system development and sustainment

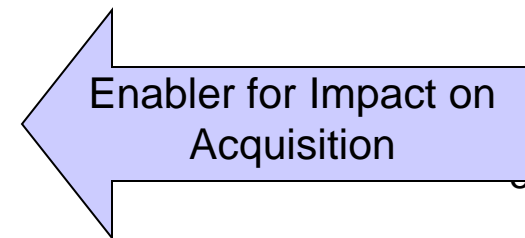
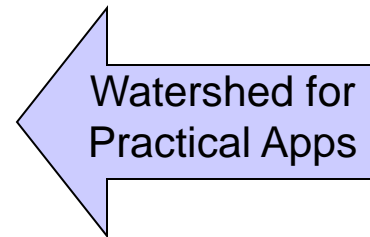
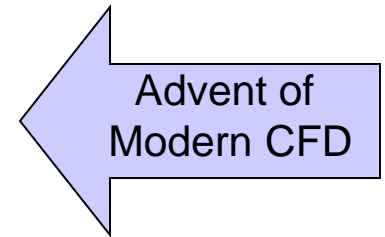




Brief History of Physics-Based Modeling in T&E

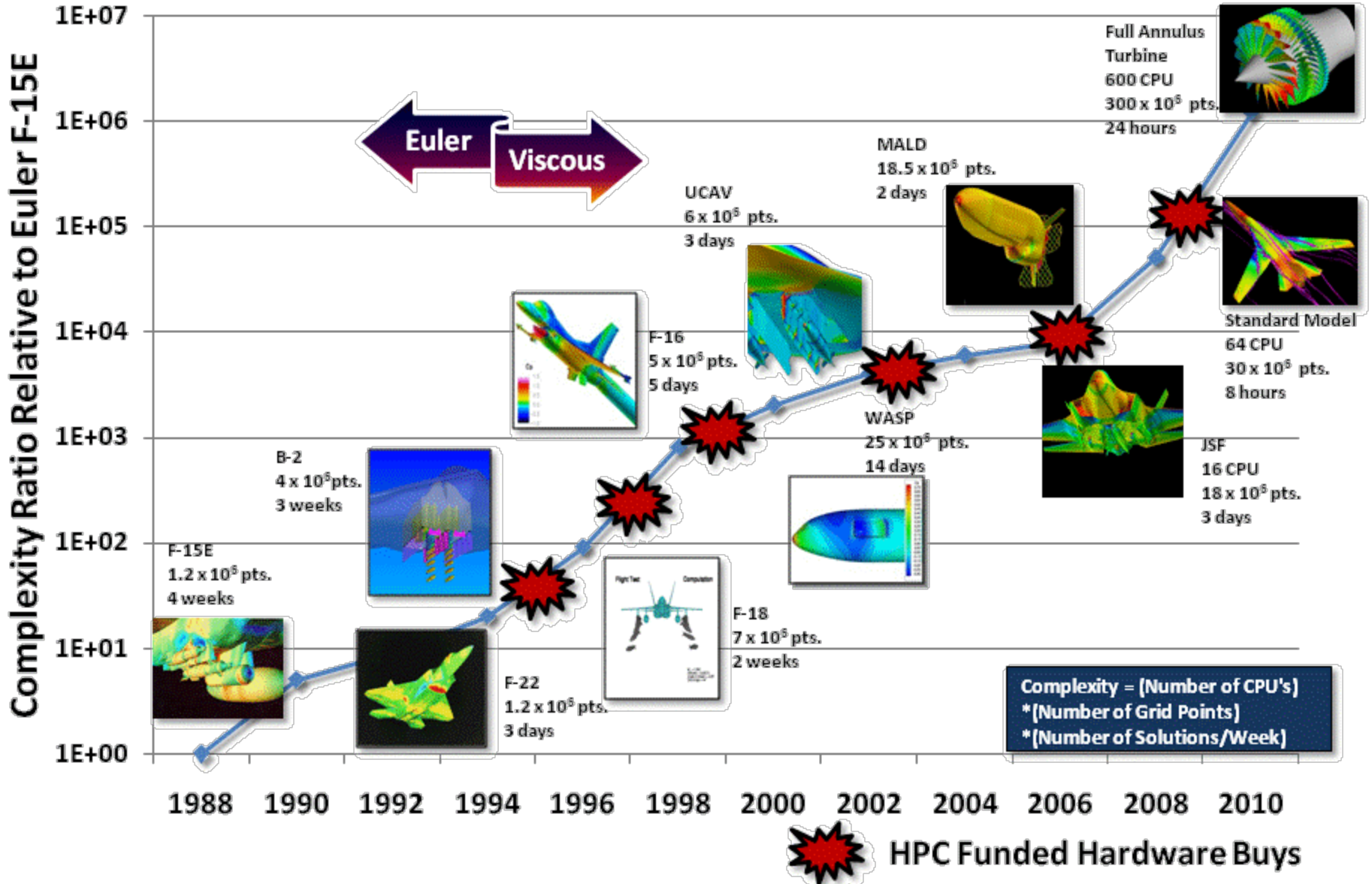


- **50's and 60's**
 - MOC, Panel Methods, and Boundary Layer Theory
 - Very limited computer capability
- **70's**
 - Non-linear small disturbance equations
 - Early IBM Mainframes
- **80's**
 - Euler, Navier-Stokes, Zonal Decomposition
 - IBM 370, Cray XMP
- **90's – today**
 - Unsteady RANS Navier-Stokes, LES
 - Clusters, Massive Parallel Processing
 - CREATE-AV Scalable Architecture





Representative Advances in Physics-Based Modeling in Support of T&E





Early EMD JSF Applications



Objective

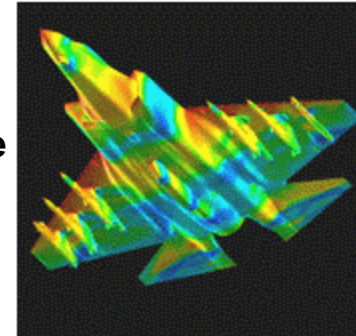
In support of early JSF developmental T&E integrated HPC with wind tunnel testing to reduce epistemic uncertainties, support design changes to improve performance, and reduce costs



Approach

In close coordination with Lockheed-Martin

- Applied high-fidelity steady and unsteady RANS CFD code
- Performed over 1000 individual computations
- Leveraged V&V with wind tunnel data to build confidence to cover regimes without data



18 million mesh points

Outcome

Used computations to increase insight and knowledge into

- Wind Tunnel Wall Effects, Reynolds Number Scaling
- Airframe Loads
- Carriage Loads
- Store Separation (Internal/External Carriage)
- Fuel Tank Design, Loads, and Jettison
- Aircraft lift fan/secondary inlet design

Impact

- Eliminated tests for high speed data
- Improved data quality and reduced risk
- Computed trajectories beyond tunnel hardware movement constraints
- Screened test configurations reducing testing costs
- Supported OEM in improving inlet performance
- Total savings = \$ Millions



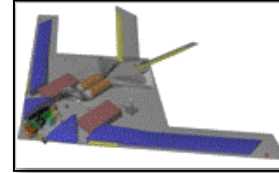
CREATE-AV

(Computational Research Engineering Acquisition Tools Environment for Air Vehicles)



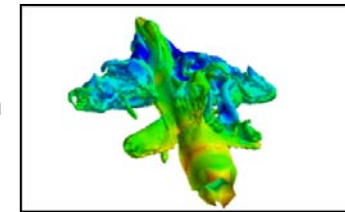
- A rapidly maturing physics-based flight system modeling architecture enabled by large scale computing
 - Development focused on impact to acquisition by embedded subject matter experts
 - Successfully delivering a family of products supporting activities from early trade studies to detailed engineering design
 - Using pilot studies to demonstrate ability to efficiently provide better physics-based design and analysis capabilities

DaVinci



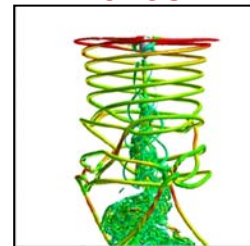
- Early engineering, design, and analysis

Kestrel



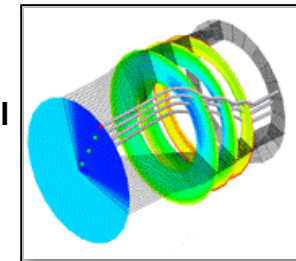
- High-fidelity, fixed wing flight system modeling

Helios



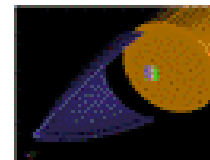
- High-fidelity, rotary wing flight system modeling

Firebolt



- Propulsion module integrated into Kestrel and Helios

Sentri



- CREATE-RF radio frequency modeling capability compatible with DaVinci



Cycle Time

Key Effectiveness Parameter



$$\text{Cycle Time} \sim \frac{\text{Workload}}{q \cdot \text{Capacity}}$$

- **Workload** – Process driven, currently ~22,000 of wind tunnel testing and 13,000 of propulsion cell testing
- **q (inverse of rework)** – Process driven, typically have 10 structural failures found in flight
- **Capacity** – Budget driven, availability x staffing x throughput

**50% reduction in wind tunnel costs equates to just a few tenths of a percent reduction in program costs –
Reducing acquisition cycle time by a month could save more than the cost of the entire wind tunnel campaign**



CREATE-AV

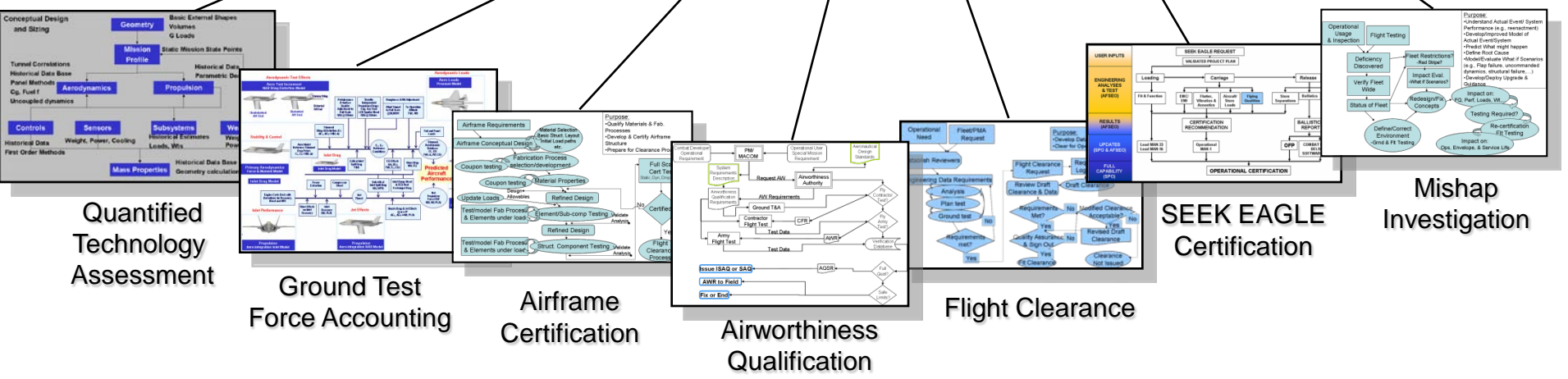
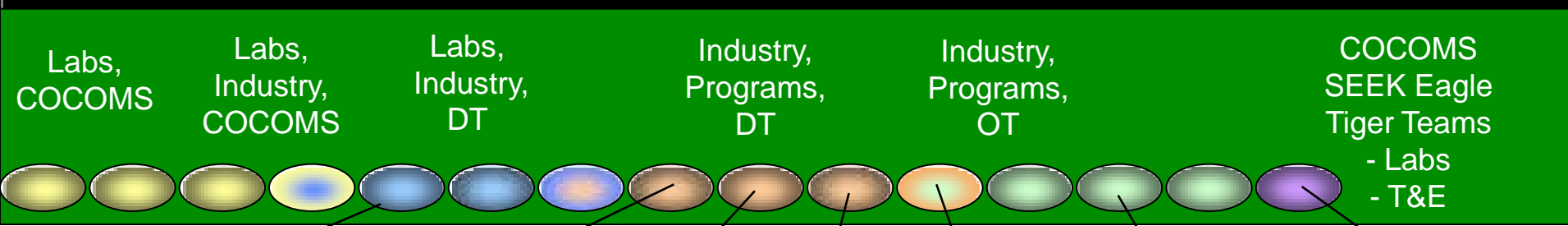
Inserting HPC Into Key Acquisition and Sustainment Processes



Acquisition Process

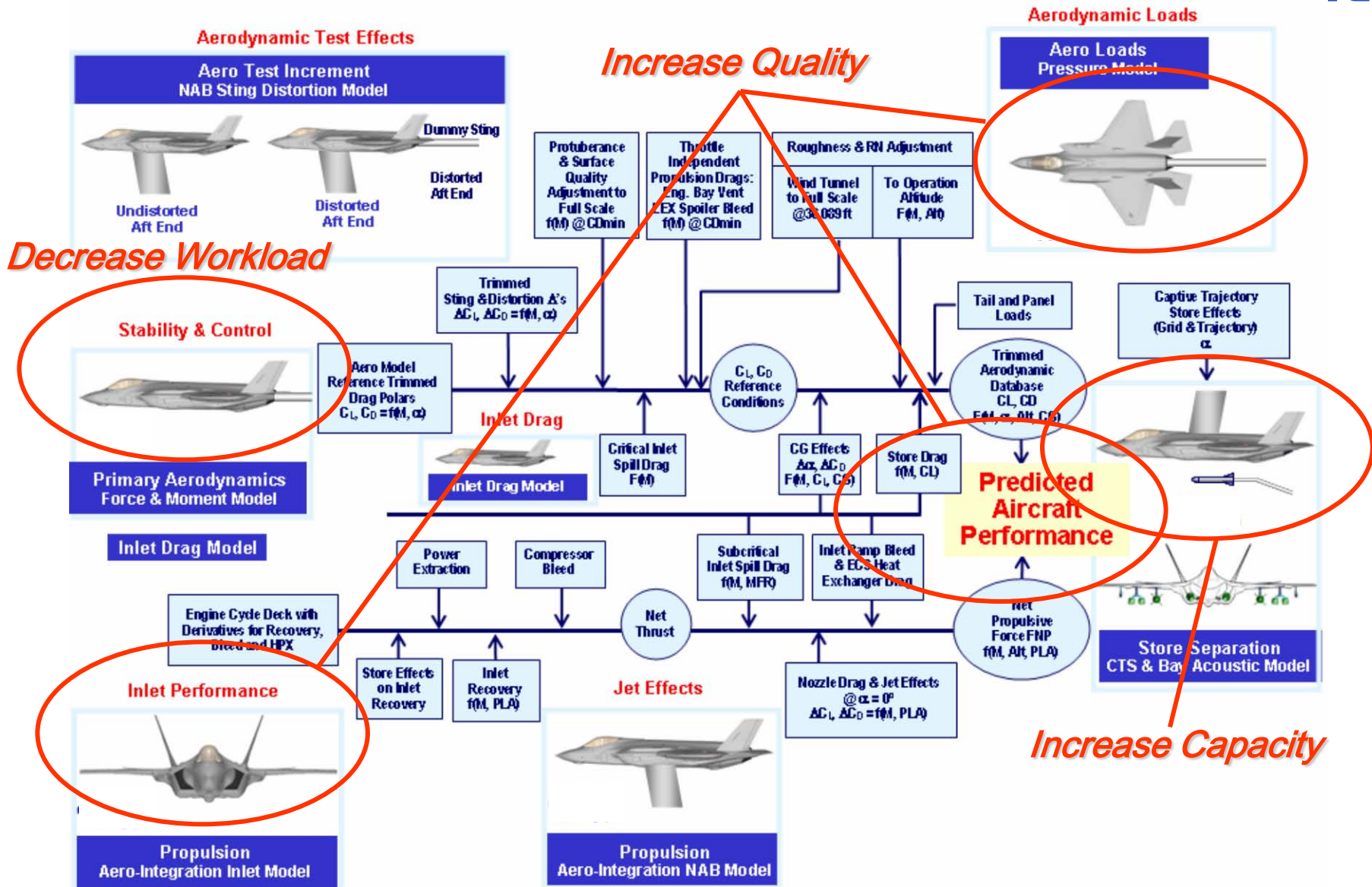


CREATE-AV – Technology Enabler to Affect Process





Aerodynamic Data Base Development Process





Reducing Workload/Increasing Capacity

Streamlining Testing at the Campaign Level

New T&E Tools + DOE



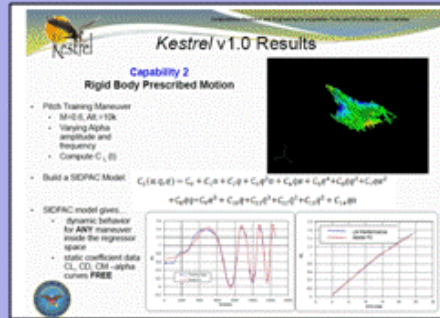
*Common Thread
System ID
Techniques*

*"Fly the Mission"
Ground Testing*



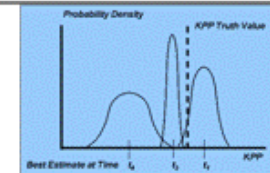
Flight Testing

*Computational Science
and Engineering Dynamic Trajectories*



*Kestrel + System ID
Response Surface*

*Estimation Theory
Quantify Effectiveness of Testing*



Using Estimation Theory' variance reduction is proportional to the effectiveness of resources used and resources applied

$$p(t_{n+1}) = p(t_n) / (1 + p(t_n) u \Delta t), \quad u = \text{resource effectiveness}$$

Or

$$u(t) = (p(t) / p(t_{n+1}) - 1) / p(t) \Delta t$$

Which can be estimated used the SEMP, TEMP, and KPP values pre- and post-test

Value of T&E

DOE

- Data Merge/Data Mine
- Response Surface Analysis
- Variance Reduction Strategy

Kraft, Edward M. "After 40 Years Why Hasn't the Computer Replaced the Wind Tunnel," The ITEA Journal of Test and Evaluation, Vol 31, pp. 329-346, September 2010.

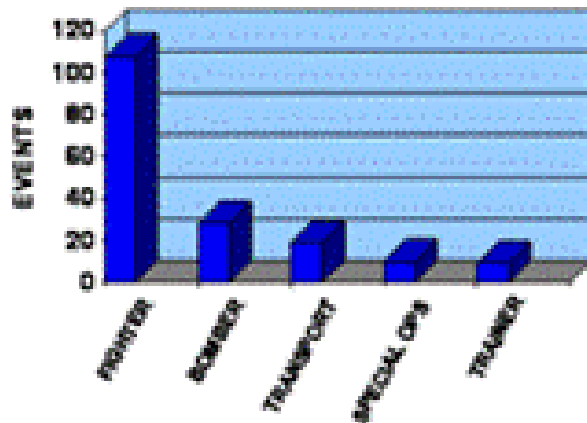


Increasing Quality

Late Structural Defect Discoveries



Structural Defect Discoveries
By Aircraft Category



When normalized by number of aircraft programs in each category, approximately 10 events per aircraft program

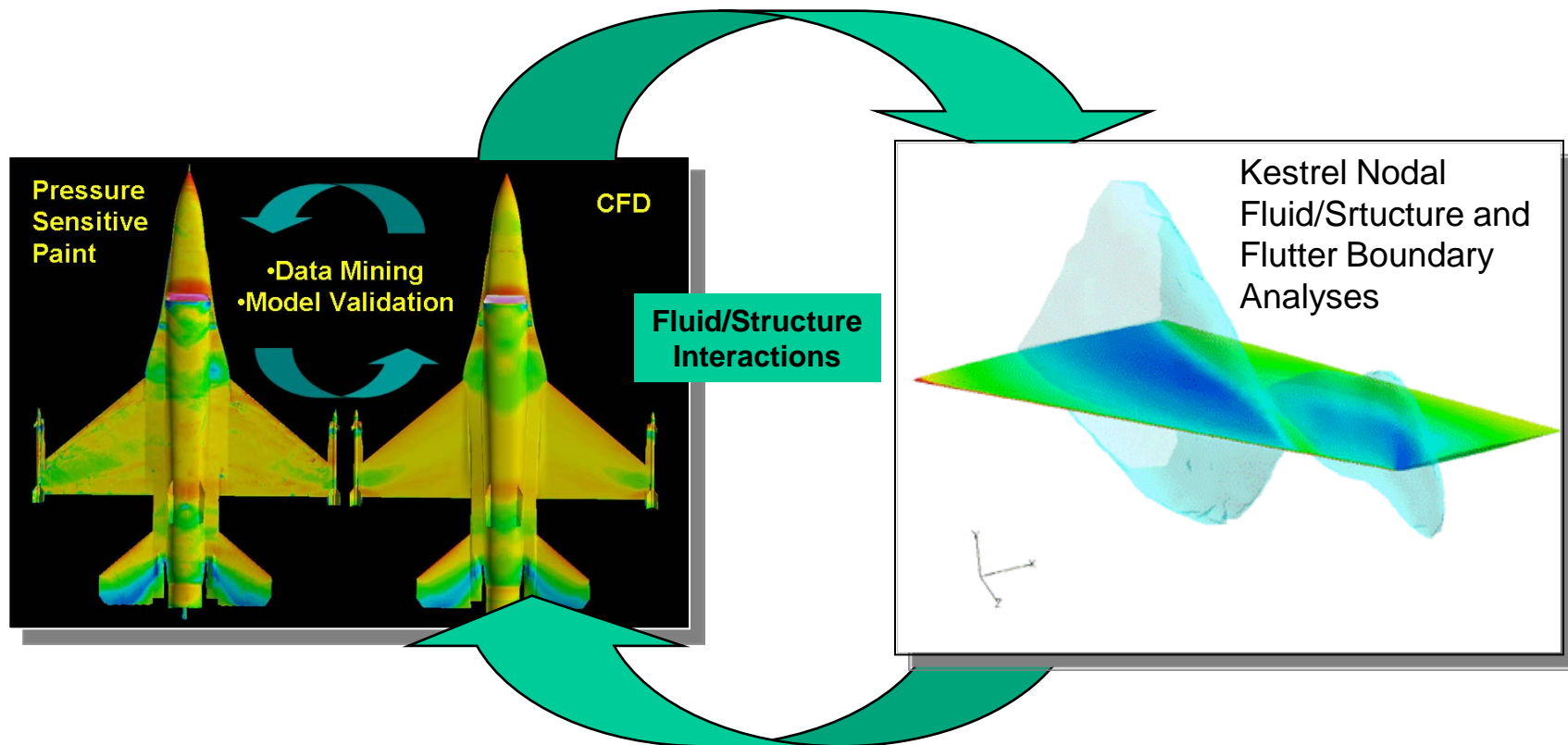
- Major Fundamental Causes
 - Inadequate Loads Analysis
 - Aero loads data base
 - Dynamic structural modeling
 - Inadequate Loads Environment
 - Inadequate estimation of non-linear local phenomena (shock, buffet, burst vortex, etc.) – gets worse for high performance aircraft

Aero load data base obtained very early in development program, loft lines not yet frozen – requires faster dynamic structural modeling capability



Reducing Late Defect Discovery

New Technologies Enhance Fluid / Structure Interactions



- Advanced PSP test technologies permit acquiring loads data more frequently during development
- Efficient nodal structural models could be updated more frequently
- Embedded finite element modeling future update to Kestrel

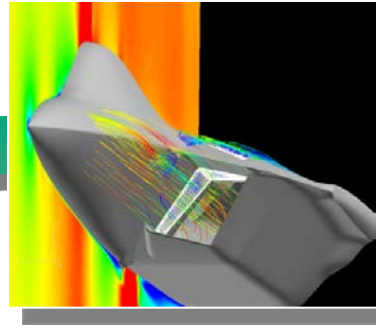
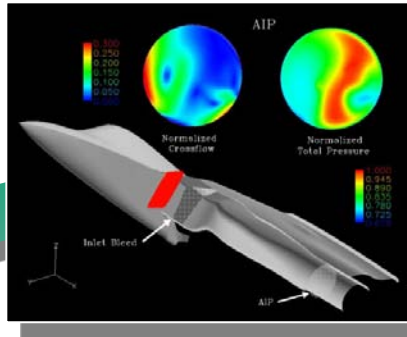


Increasing Quality

Early Airframe / Propulsion Integration



Inlet Design

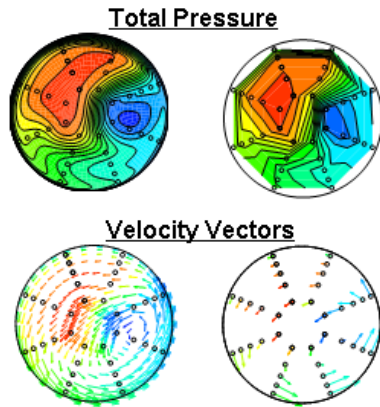


Wind Tunnel Validation

- Improved Inlet Performance
- Focused Testing
- Lowered Risk for Advanced Inlet Concepts

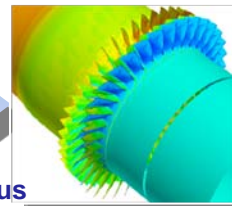


Embedded in Maneuvering A/C



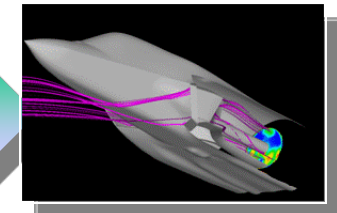
Firebolt

Full Annulus Modeling



+ Kestrel

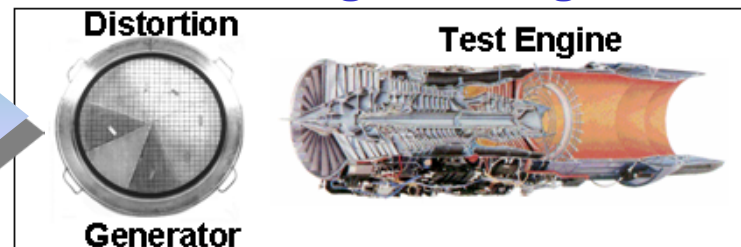
Engine Design



CFD Resolution

Equivalent 40-Probe Rake

Data Merging / Data Mining



Engine Testing



Summary



- **High-fidelity physics based modeling has been an integral part of aeronautical T&E for over 25 years**
- **CREATE-AV is an enabler to accelerate the acquisition process**
- **Focus needs to be on processes changes, not just the science of high performance computing**